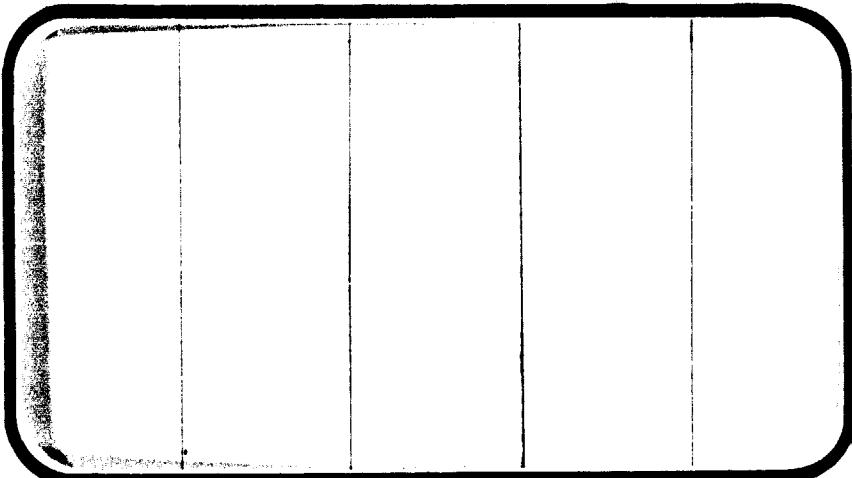




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(NASA-CR-141511) REENTRY AERODYNAMIC
CHARACTERISTICS OF A SPACE SHUTTLE SOLID
ROCKET BOOSTER (MSFC MODEL 454) AT HIGH
ANGLES OF ATTACK AND HIGH MACH NUMBER IN THE
NASA/LANGLEY FOUR-FOOT UNITARY PLAN WIND

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SPACE SHUTTLE

AEROTHERMODYNAMIC DATA REPORT

JOHNSON SPACE CENTER
HOUSTON, TEXAS

DATA MANAGEMENT services
SPACE DIVISION  CHRYSLER
CORPORATION

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REENTRY AERODYNAMIC CHARACTERISTICS OF A
SPACE SHUTTLE SOLID ROCKET BOOSTER (MSFC MODEL
454) AT HIGH ANGLES OF ATTACK AND HIGH MACH
NUMBER IN THE NASA/LANGLEY FOUR-FOOT UNITARY
PLAN WIND TUNNEL (SA25F)

by

J. Johnson, NASA/MSFC
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Prepared Under NASA Contract Number NAS9-13247

by

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for

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Johnson Space Center
National Aeronautics and Space Administration
Houston, Texas

WIND TUNNEL TEST SPECIFICS:

Test Number: LaRC 4' UPWT 1087
NASA Series No.: SA25F
Model Number: MSFC 454
Test Dates: March 4 - 11, 1974
Occupancy Hours: 30

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Chrysler Corporation Space Division assumes no responsibility for the
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REENTRY AERODYNAMIC CHARACTERISTICS OF A
SPACE SHUTTLE SOLID ROCKET BOOSTER (MSFC MODEL
454) AT HIGH ANGLES OF ATTACK AND HIGH MACH
NUMBER IN THE NASA/LANGLEY FOUR-FOOT UNITARY PLAN WIND
TUNNEL (SA25F)

By

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ABSTRACT

A force test of a 2.112 percent scale Space Shuttle Solid Rocket Booster (SRB), MSFC Model 454, was conducted in test section #2 of the Langley Research Center's Unitary Plan Wind Tunnel. This test, UPWT 1087, occupied the tunnel for 30 hours between March 4 and March 11, 1974. Sixteen (16) runs (pitch polars) were performed over an angle of attack range from 144 through 179 degrees. Test Mach numbers were 2.30, 2.70, 2.96, 3.48, 4.00 and 4.63. The first three Mach numbers had a test Reynolds number of 1.5 million per foot. The remaining three were at 2.0 million per foot. The model was tested in the following configurations:

1. SRB without external protuberances
2. SRB with an electrical tunnel and a SRB/ET thrust attachment structure.

Schlieren photographs were taken during testing of the first configuration. The second configuration was tested at roll angles of 45, 90 and 135 degrees.

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PLOTTED COEFFICIENTS SCHEDULE

(A) CNM, CA, CLMM, CYM, CYNM, CBL Versus ALPHA

NOMENCLATURE

| <u>SYMBOL</u> | <u>PLOT SYMBOL</u> | <u>DEFINITION</u> | <u>UNITS</u> |
|---------------|--------------------|---|------------------|
| A_b | | base area; cross-sectional area of the cylindrical section of the model | in. ² |
| A_c | | cavity area, area of the opening required for the balance and sting | in. ² |
| BMC | BMC | Balance Moment Center | |
| b_{ref} | BREF | reference span (diameter of the cylindrical section of the model) | in. |
| ℓ_{body} | | length of the body | in. |
| ℓ_{ref} | LREF | reference length (diameter of the cylindrical section of the model) | in. |
| M | MACH | Mach number | |
| MRP | MRP | Moment Reference Point (located by XMRP, YMRP, and ZMRP) | in. |
| p_{bi} | | base pressures | psi |
| p_t | | free stream total pressure | psi |
| p_∞ | | free stream static pressure | psi |
| q_∞ | Q(PSF) | free stream dynamic pressure | psi |
| R_N | | Reynolds number based on ℓ_{ref} | |
| R_N/ft | RN/L | Reynolds number per unit length | per ft. |
| S_{ref} | SRFF | reference area (cross sectional area of the cylindrical section of the model) | in. ² |
| T_t | | tunnel total temperature | °F |
| C_{pc} | CPC | pressure coefficient of balance cavity | |

NOMENCLATURE (Continued)

| <u>SYMBOL</u> | <u>PLOT SYMBOL</u> | <u>DEFINITION</u> | <u>UNITS</u> |
|------------------------|------------------------|---|--------------|
| X, Y, Z | | body axes system coordinates (for an airplane, the X, Z-plane is the plane of symmetry, the origin of the axes system is the center of gravity or any other convenient point, and the X axis is the airplane longitudinal axis) | in. |
| $x_{c.g.}$ | | distance of center of gravity from nose of body | in. |
| X_m, Y_m, Z_m | | missile axes (see text) | in. |
| XMRP, YMRP, ZMRP | XMRP, YMRP, ZMRP | Abbreviations for the location of the Moment Reference Point in the missile axis system | in. |
| α_T | ALPHA | total angle-of-attack, angle between the X_m -axis and a vector in the direction of the relative wind | degrees |
| ϕ | PHI | roll angle; i.e., angle between the missile Y_m -axis and the body Y-axis (from a pilot's viewpoint in an airplane, a positive roll angle is a clockwise rotation). | degrees |
| C_A | CA | total axial force coefficient in the body axis system | |
| C_{A_b} | CAB | base axial force coefficient (same in both missile and body axis systems) | |
| $C_{A_{f_m}}$ | CAF | forebody axial force coefficient, $C_A - C_{A_b}$ | |
| C_{A_m} | CA | total axial force coefficient in the missile axis system, $F_{A_m}/q_\infty S_{ref}$ | |
| C_l | CBL | rolling moment coefficient in the body axis system | |
| C_{l_m} | CEL | rolling moment coefficient in the missile axis system, $M_{y_m}/q_\infty S_{ref} l_{ref}$ | |

NOMENCLATURE (Continued)

| <u>SYMBOL</u> | <u>PLOT SYMBOL</u> | <u>DEFINITION</u> | <u>UNITS</u> |
|---------------|--------------------|--|--------------|
| C_m | CLM | pitching moment coefficient in the body axis system | |
| C_{m_m} | CLMM | pitching moment coefficient in the missile axis system, $M_y/m / q_\infty S_{ref} \cdot ref$ | |
| C_N | CN | normal force coefficient in the body axis system | |
| C_{N_m} | CNM | normal force coefficient in the missile axis system, $F_N/m / q_\infty S_{ref}$ | |
| C_n | CYN | yawing moment coefficient in the body axis system | |
| C_{n_m} | CYNM | yawing moment coefficient in the missile axis system, $M_z/m / q_\infty S_{ref} \cdot ref$ | |
| $C_{P_{bl}}$ | | base pressure coefficient: $\frac{P_{bl}-P}{q_\infty}$ | |
| C_Y | CY | side force coefficient in the body axis system | |
| C_{Y_m} | CYM | side force coefficient in the missile axis system, $F_y/m / q_\infty S_{ref}$ | |
| x_{cp}/l | ACP/L | center of pressure location in fraction of body length from nose; | |
| | | $\left[\frac{x_{c.p.g.}}{\text{body}} - \frac{C_{m_m}}{C_{N_m}} \cdot \frac{\text{ref}}{\text{body}} \right]$ | |
| F_{Y_m} | SF | side force in the missile axis system, positive in the positive direction of Y_m | lb |
| F_{A_m} | AF | total axial force in the missile axis system, positive in the negative direction of X_m | lb |
| F_{N_m} | NF | normal force in the missile axis system, positive in the negative direction of Z_m | lb |

NOMENCLATURE (Concluded)

| <u>SYMBOL</u> | <u>PLOT SYMBOL</u> | <u>DEFINITION</u> | <u>UNITS</u> |
|---------------|--------------------|--|--------------|
| M_{X_m} | RM | rolling moment in the missile axis system; i.e., moment about the X_m -axis (a positive rolling moment tends to rotate the positive Y_m -axis toward the positive Z_m -axis) | in.-lb |
| M_{Y_m} | PM | pitching moment in the missile axis system; i.e., moment about the Y_m -axis (a positive pitching moment tends to rotate the positive Z_m -axis toward the positive X_m -axis) | in.-lb |
| M_{Z_m} | YM | yawing moment in the missile axis system; i.e., moment about the Z_m -axis (a positive yawing moment tends to rotate the positive X_m -axis toward the positive Y_m -axis) | in.-lb |
| β | BETA | angle of sideslip | deg. |
| C_L | CL | lift coefficient (stability-axis system) | |
| C_D | CD | drag coefficient (stability-axis system) | |

SUBSCRIPTS

| | |
|----------|---|
| b | base |
| c | cavity |
| c.g. | center of gravity |
| i | identifies the location of the base pressure measurements |
| m | missile axis system |
| ref | reference conditions |
| t | total conditions |
| ∞ | free stream conditions |

INTRODUCTION

In a continuing effort to determine the aerodynamic characteristics of a space shuttle solid rocket booster (SRB) during reentry, a test was conducted at the Langley Research Center 4 foot Unitary Plan wind Tunnel. Static stability coefficients were obtained at high Mach numbers and high angles of attack. This report describes the test that was performed and presents the data that were obtained.

Wind tunnel conditions during this test were as follows: Mach numbers were 2.3, 2.7, 3.0, 3.5, 4.0, and 4.63; angle of attack range was 144 through 179 degrees; and Reynolds numbers were 1.5 and 2.0 million per foot.

The two model configurations investigated during this test were:

- o SRB without external protuberances
- o SRB with an electrical tunnel and a SRB/ET thrust attachment structure

MODEL AND SUPPORT HARDWARE

MODEL

The model, MSFC wind tunnel model 454, was a 0.02112 scale model of the 142-inch diameter Space Shuttle solid rocket booster. Details of the model are presented in Figures 2, 3, and 4 and Table III. Figure 2 presents the major dimensions of the model. Figure 3 presents the dimensions and location of a scaled attachment ring. Figure 4 presents the dimensions and location of the electrical tunnel and external tank thrust attachment structure protuberances which were attached to the model during selected portions of this test.

The model was made in seven sections. These sections, as identified in Figure 5, are a nose, a strake ring, two fill rings, a balance body, a balance body end section, and a tail section. The model was made in these seven sections to facilitate testing at all angles of attack from 0 to 180 degrees. The features that were utilized for this test were:

- o The two fill rings were assembled upstream of the balance to position the anticipated center of pressure as close as possible to the balance center.
- o The tail section was mounted upstream and the nose and strake ring sections were mounted downstream to simulate high angles of attack approaching 180 degrees. A truncated cone nose section (Figure 2) allowed the passage of the sting.

The model is designed so that roll angles of 0, 45, 90, 135, 180, 225, 270, and 315 degrees can be simulated. These roll angles can be achieved by rolling the nose and tail sections relative to the balance body. The nose and tail sections can be rolled in 22 1/2 degree increments; however, screw holes in the balance body and sting outlet section for the electrical tunnel are only provided every 45 degrees. Therefore, roll angle changes are limited to changes in 45 degree increments. The roll angle sign convention for the SRB protuberances is shown in Figure 4.

For more information on the capabilities of this model, see Reference 2.

SUPPORT HARDWARE

The support hardware was provided by the test facility and included:

Knuckle #4 (angle of attack mechanism)

Knuckle offset (adjusted to 7 1/2 inches)

1.5° coupling

Sting UT 30

Balance #834

Knuckle #4 was the normal angle of attack mechanism for the facility and had a range from -14 to +21 degrees. The action of the knuckle produced a fore and aft translation as well as an angle of attack change.

The center of rotation produced by knuckle #4 was behind the model. With a model as long as the one used, and at angles at which it was tested, a knuckle offset was necessary. This offset placed the centerline of the sting 7 1/2 inches below the centerline of knuckle #4.

To adjust the -14 to +21 degrees range of knuckle #4 to angles more desirable for this test, a 15 degree coupling was used. This gave balance angles of 1 to 36 degrees which for a tail forward model gave angles of attack of 179 to 144 degrees.

CONFIGURATIONS INVESTIGATED

The run schedule, i.e., data set collation sheet, for this test, LaRC UPWT 1087, is shown in Table II. This table contains the data set identifiers for the test and gives the nominal conditions at which various configurations were tested. These conditions are angle of attack (α_T), roll angle (ϕ), and Mach number.

Configuration SRB w/attach ring was a 2.112 percent scale model of a 142 inch diameter SRB configuration, less electrical tunnel and external tank thrust attachment structure (Figures 2 and 3).

The configuration SRB w/attach ring, electrical tunnel, and thrust attachment, was formed by adding the two scaled protuberances to the previous configuration as shown in Figure 4. This configuration, tested at roll angles of 45, 90, and 135 degrees, was created by placing the scaled electrical tunnel and SRB/ET thrust attach structure on the model at the appropriate roll angle. Investigations were made over an alpha range from 144 to 179 degrees at Mach numbers of 2.3 through 4.63. A description of the individual model components is given in Table III.

TEST FACILITY DESCRIPTION

This wind tunnel test was conducted in the Unitary Plan Wind Tunnel (UPWT) at the Langley Research Center. The UPWT is a two leg, closed return, continuous run type wind tunnel. The test sections are four by four feet in cross section and seven feet long. The Mach number range is from 1.5 to 3.0 in the low Mach number test section and 2.3 to 4.63 in the test section used during this test. This high Mach number test section has a Reynolds number capability as high as 7.7×10^6 per foot. Additional details about this facility are presented in Reference 1.

DATA ACQUISITION AND REDUCTION

The parameters measured and recorded during this test were:

- o Wind tunnel conditions (P_∞ , P_t , T_t)
- o Six-component force and moment data
- o Sting attitude
- o Cavity pressure

Tunnel conditions were used to calculate the Mach number, the dynamic pressure, and the Reynolds number (Table I); the six-component force and moment data were used to calculate static stability coefficients; the sting attitude, nominal model attitude, and deflection calibrations were used to calculate the model angle of attack; and the cavity pressures were used to calculate cavity pressure coefficients.

Figure 6 shows the location of the cavity pressure tubes. A tabulation of the cavity pressure coefficients ($C_{P_{ci}}$) are included in the appendix to this report.

As stated above, the six component force and moment data were used to calculate six-component static stability coefficients. These data were measured with Langley Research Center Balance #834. The rated capacities of this balance are listed in Table I. The six coefficients, C_{A_m} , C_{C_m} , C_{M_m} , C_{N_m} , C_{R_m} , and C_{Y_m} , are coefficients in the missile axis system.

The missile axis system (X_M, Y_M, Z_M) is a non-rolling body axis system that is frequently used in wind tunnel tests and studies of missile flight dynamics. It is a system of axes that never rotates about the missile or model longitudinal axis. The orientations of the missile axes coefficients

are defined in Figure 1. The missile axis system is identical with the body axis system at zero roll angle.

Six-component static aerodynamic coefficients in the missile axis system may be converted to coefficients in the body axis system with the following six equations:

$$C_A = C_{A_m}$$

$$C_N = C_{N_m} \cos \phi + C_{Y_m} \sin \phi$$

$$C_Y = -C_{N_m} \sin \phi + C_{Y_m} \cos \phi$$

$$C_L = C_{L_m}$$

$$C_m = C_{m_m} \cos \phi + C_{n_m} \sin \phi$$

$$C_n = -C_{m_m} \sin \phi + C_{n_m} \cos \phi$$

The following reference dimensions were used to calculate the static stability coefficients:

| <u>Parameter</u> | <u>Full Scale</u> | <u>Model Scale</u> |
|---|------------------------|------------------------|
| Reference Area (S_{ref}) based on body cross section | 109.98 ft ² | 7.069 in. ² |
| Reference Length (ℓ_{ref}) = (b_{ref}) = model diameter | 142 in. | 3.000 in. |
| Moment Reference Center (from body nose) | | |
| *XMRP (1.134 inches upstream of balance center) | 986.97 in. | 20.834 in. |
| XMRP | 0 | 0 |
| ZMRP | 0 | 0 |

*Note: XMRP (56.69% of total length without portion of nose removed,
measured from nose tip)

The force and moment data were corrected for model weight tares but tunnel flow angularity was assumed to be zero.

Schlieren photographs were made during this test.

DATA PRESENTATION

Data are presented in two forms: (1) aerodynamic static stability coefficients are plotted as a function of angle-of-attack and (2) data tables are presented that include six static stability coefficients, two cavity pressure coefficients, wind tunnel flow conditions, and model attitude.

Data Plots

The plots of the six static stability coefficients are presented in the following groups:

- o Static stability characteristics of SRB (with attachment ring).
- o Static stability characteristics of SRB (with attach ring, electrical tunnel, and thrust attachment) for various roll angles.

Table IV presents, for each configuration or comparison of configurations, the roll angles and the Mach numbers for which data are available.

Data Tables

Data tables, identified as tabulated source data, are presented for each of the 16 runs that were made during this test, with the exception of run 12. Run 12 was found to have been made at tunnel conditions other than what was desired. Facility personnel elected to delete this run. The data are presented in the order of data set number. Each data set contains information that describes the model configuration, the model attitude, the tunnel flow conditions, and the model reference dimensions. Each data set contains a listing of the six static aerodynamic stability coefficients and two cavity pressure coefficients ($C_{p_{c1}}$). Zeros appeared in the

cavity pressure columns for runs 1 through 6. Equipment problems during these runs resulted in unreliable pressure data. Therefore, these pressure coefficients are not shown in the tabulated listing.

If a cavity pressure correction is desired, the cavity area to be used is:

$$A_C = 3.976 \text{ in.}^2$$

REFERENCES

1. "Unitary Plan Wind Tunnel Facilities," National Advisory Committee for Aeronautics; 1956.
2. DMS-DR-2161, "Aerodynamic Characteristics of a 1/2-Inch Diameter Solid Rocket Booster (Configuration 139)," Radford, W. D., Johnson, J. D., Burstadt. P. L.,

Table I.

| TEST: LARC UPWT 1087 | | DATE: March 1974 | | |
|------------------------------------|----------------------------|-----------------------------------|---|--------------------------------------|
| TEST CONDITIONS | | | | |
| MACH NUMBER | REYNOLDS NUMBER (per foot) | DYNAMIC PRESSURE (pounds/sq.inch) | STAGNATION TEMPERATURE (degrees Fahrenheit) | STAGNATION PRESSURE (pounds/sq.inch) |
| 2.30 | 1.5×10^6 | 2.36 | 155 | 7.97 |
| 2.70 | | 2.16 | 155 | 9.85 |
| 2.96 | ↓ | 2.01 | 155 | 11.34 |
| 3.48 | 2.0×10^6 | 2.26 | 150 | 19.76 |
| 4.00 | | 2.02 | 185 | 27.39 |
| 4.63 | ↓ | 1.62 | 185 | 36.69 |
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| BALANCE UTILIZED: <u>LARC #834</u> | | | | |
| CAPACITY: | | ACCURACY: | COEFFICIENT TOLERANCE: | |
| NF | <u>300 lbs</u> | _____ | _____ | |
| SF | <u>200 lbs</u> | _____ | _____ | |
| AF | <u>60 lbs</u> | _____ | _____ | |
| PM | <u>1000 in.-lbs</u> | _____ | _____ | |
| RM | <u>100 in.-lbs</u> | _____ | _____ | |
| YM | <u>400 in.-lbs</u> | _____ | _____ | |
| COMMENTS: | | | | |

Table III.
MODEL COMPONENT DIMENSIONS

MODEL COMPONENT: Nose

GENERAL DESCRIPTION: 142-inch diameter SRB nose, cone angle is 18° with a spherical radius nose cap. (This nose was truncated to allow passage of the sting when the model was nose-mounted for testing at 144 degrees $\leq \alpha_s \leq 179$ degrees)

MODEL DRAWING NUMBER: MSFC #80M42621
MSFC #80M42622

| <u>DIMENSIONS:</u> | <u>THEORETICAL</u> | | <u>ACTUAL MEASURED</u> |
|--|------------------------------|------------------------------|------------------------|
| | <u>FULL-SCALE</u> | <u>MODEL SCALE</u> | <u>MODEL SCALE</u> |
| Length | <u>188 in.</u> | <u>3.971 in.</u> | <u>3.892 in.</u> |
| Max. Width | <u>142 in.</u> | <u>3.000 in.</u> | <u>3.000 in.</u> |
| Max. Depth | <u>142 in.</u> | <u>3.000 in.</u> | <u>3.000 in.</u> |
| Fineness Ratio | <u>1.32</u> | <u>1.32</u> | <u>1.297</u> |
| Area | | | |
| Max. Cross-Sectional | <u>109.97 ft²</u> | <u>7.069 in.²</u> | _____ |
| Planform | _____ | _____ | _____ |
| Wetted | _____ | _____ | _____ |
| Base | <u>109.97 ft²</u> | <u>7.069 in.²</u> | _____ |
| Length (when truncated for sting mounting) | _____ | <u>1.130 in.</u> | <u>1.130 in.</u> |

Table III. (Continued)

MODEL COMPONENT: Body

GENERAL DESCRIPTION: 142-inch diameter SRB body for SRB configuration 139.

MODEL DRAWING NUMBER: 80M42621, 80M42623, 80M32580, 80M51331, 80M42646, 80M32581
80M42626

| <u>DIMENSIONS:</u> | <u>THEORETICAL</u> | | <u>ACTUAL MEASURED</u> |
|----------------------|------------------------------|------------------------------|------------------------|
| | <u>FULL-SCALE</u> | <u>MODEL SCALE</u> | <u>MODEL SCALE</u> |
| Length | <u>1407.8 in.</u> | <u>29.734 in.</u> | <u>29.757 in.</u> |
| Max. Width | <u>142 in.</u> | <u>3.000 in.</u> | <u>3.001 in.</u> |
| Max. Depth | <u>142 in.</u> | <u>3.000 in.</u> | <u>3.001 in.</u> |
| Fineness Ratio | <u>9.91</u> | <u>9.91</u> | <u>9.915</u> |
| Area | | | |
| Max. Cross-Sectional | <u>109.98 ft²</u> | <u>7.069 in.²</u> | |
| Planform | | | |
| Wetted | | | |
| Base | <u>109.97 ft²</u> | <u>7.069 in.²</u> | |

Table III.(Continued)

MODEL COMPONENT: Engine/Shroud

GENERAL DESCRIPTION: 142-inch diameter SRB engine shroud/nozzle combination for SRB configuration 139.

MODEL DRAWING NUMBER: 80M42626, 80M32613

| <u>DIMENSIONS:</u> | <u>THEORETICAL</u> | | <u>ACTUAL MEASURED</u> |
|---------------------------|------------------------------|-------------------------------|-------------------------------|
| | <u>FULL-SCALE</u> | <u>MODEL SCALE</u> | <u>MODEL SCALE</u> |
| <u>Engine Shroud</u> | | | |
| Length | <u>93 in.</u> | <u>1.964 in.</u> | <u>1.975 in.</u> |
| Max. Width | <u>192 in.</u> | <u>4.055 in.</u> | <u>4.062 in.</u> |
| Max. Depth | <u>192 in.</u> | <u>4.055 in.</u> | <u>4.062 in.</u> |
| Max. Cross Sectional Area | <u>201.1 ft²</u> | <u>12.914 in.²</u> | <u>12.959 in.²</u> |
| <u>Engine Nozzle</u> | | | |
| Length | <u>52 in.</u> | <u>1.102 in.</u> | <u>1.095 in.</u> |
| Max. Width | <u>141.7 in.</u> | <u>2.993 in.</u> | <u>2.994 in.</u> |
| Max. Depth | <u>141.7 in.</u> | <u>2.993 in.</u> | <u>2.994 in.</u> |
| Max. Cross Sectional Area | <u>109.52 ft²</u> | <u>7.040 in.²</u> | <u>7.040 in.²</u> |

Table III. (Continued)

MODEL COMPONENT: Attachment Ring

GENERAL DESCRIPTION: An attachment ring (used to attach the SRB to the ET)

is located 27.773 inches model scale (1315 inches full scale) from the nose of
the vehicle.

MODEL DRAWING NUMBER: 80M32547, 80M32582

| <u>DIMENSIONS:</u> | <u>THEORETICAL</u> | | <u>ACTUAL MEASURED</u> |
|----------------------|--------------------|--------------------|------------------------|
| | <u>FULL-SCALE</u> | <u>MODEL SCALE</u> | <u>MODEL SCALE</u> |
| Length | | | |
| Max. Width | <u>10.98 in.</u> | <u>0.232 in.</u> | |
| Max. Depth | <u>9.99 in.</u> | <u>0.211 in.</u> | |
| Fineness Ratio | | | |
| Area | | | |
| Max. Cross-Sectional | | | |
| Planform | | | |
| Wetted | | | |
| Base | | | |

Table III.(Continued)

MODEL COMPONENT: Electrical Tunnel

GENERAL DESCRIPTION: The electrical tunnel runs along the outside the SRB

tank to protect the various electrical cables from aerodynamic loading.

MODEL DRAWING NUMBER: 80M42642

| <u>DIMENSIONS:</u> | <u>THEORETICAL</u> | | <u>ACTUAL MEASURED</u> |
|----------------------|--------------------|--------------------|------------------------|
| | <u>FULL-SCALE</u> | <u>MODEL SCALE</u> | <u>MODEL SCALE</u> |
| Length | <u>1274 in.</u> | <u>.26.4 in.</u> | <u></u> |
| Max. Width | <u>13 in.</u> | <u>0.275 in.</u> | <u></u> |
| Max. Depth | <u>6 in.</u> | <u>0.127 in.</u> | <u></u> |
| Fineness Ratio | <u></u> | <u></u> | <u></u> |
| Area | <u></u> | <u></u> | <u></u> |
| Max. Cross-Sectional | <u></u> | <u></u> | <u></u> |
| Planform | <u></u> | <u></u> | <u></u> |
| Wetted | <u></u> | <u></u> | <u></u> |
| Base | <u></u> | <u></u> | <u></u> |

Table III. (Concluded)

MODEL COMPONENT: SRB/ET Thrust Attachment Structure

GENERAL DESCRIPTION: This structure is mounted aft the intersection of nose
and body and is used to attach the SRB to the ET.

MODEL DRAWING NUMBER: 80M42641

| <u>DIMENSIONS:</u> | <u>THEORETICAL</u> | | <u>ACTUAL MEASURED</u> |
|----------------------|--------------------|--------------------|------------------------|
| | <u>FULL-SCALE</u> | <u>MODEL SCALE</u> | <u>MODEL SCALE</u> |
| Length | <u>~ 47.3 in.</u> | <u>~ 1.000 in.</u> | |
| Max. Width | <u>~ 129.9 in.</u> | <u>~ 2.744 in.</u> | |
| Max. Depth | <u>~ 7.4 in.</u> | <u>~ 0.157 in.</u> | |
| Fineness Ratio | | | |
| Area | | | |
| Max. Cross-Sectional | | | |
| Planform | | | |
| Wetted | | | |
| Base | | | |

NORTHROP SERVICES, INC.

M-9230-74-234

Table IV. PLOT SUMMARY

| INVESTIGATION | ROLL ANGLE | | | | MACH NUMBER | | | | | |
|--|------------|----|----|-----|-------------|-----|-----|-----|-----|------|
| | 0 | 45 | 90 | 135 | 2.3 | 2.7 | 3.0 | 3.5 | 4.0 | 4.63 |
| Static stability characteristics of SRB with attachment ring | X | | | | X | X | X | X | X | X |
| Effect of roll on static stability characteristics of SRB (with protuberances) | | X | X | X | | X | | | X | X |

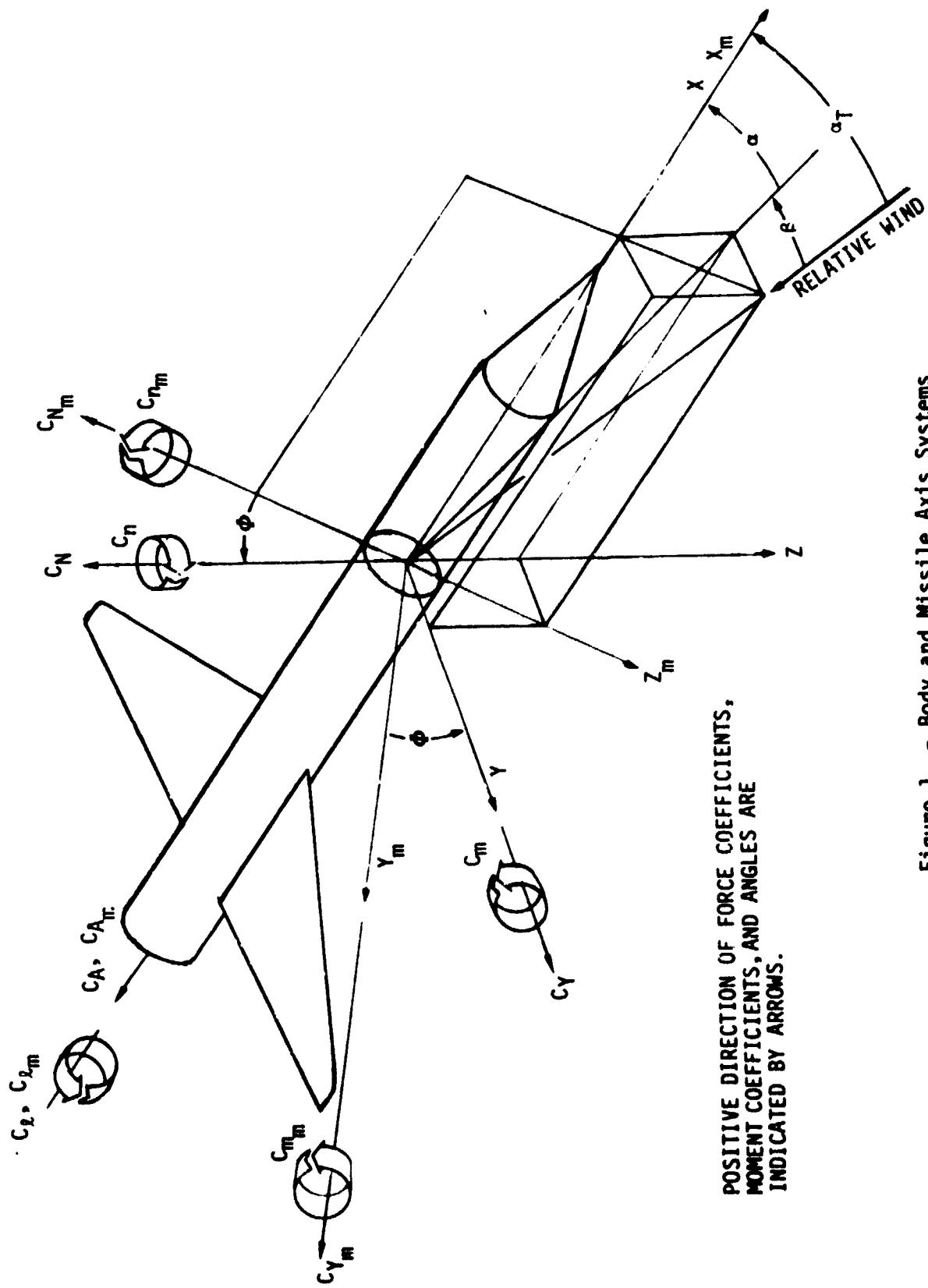


Figure 1. - Body and Missile Axis Systems

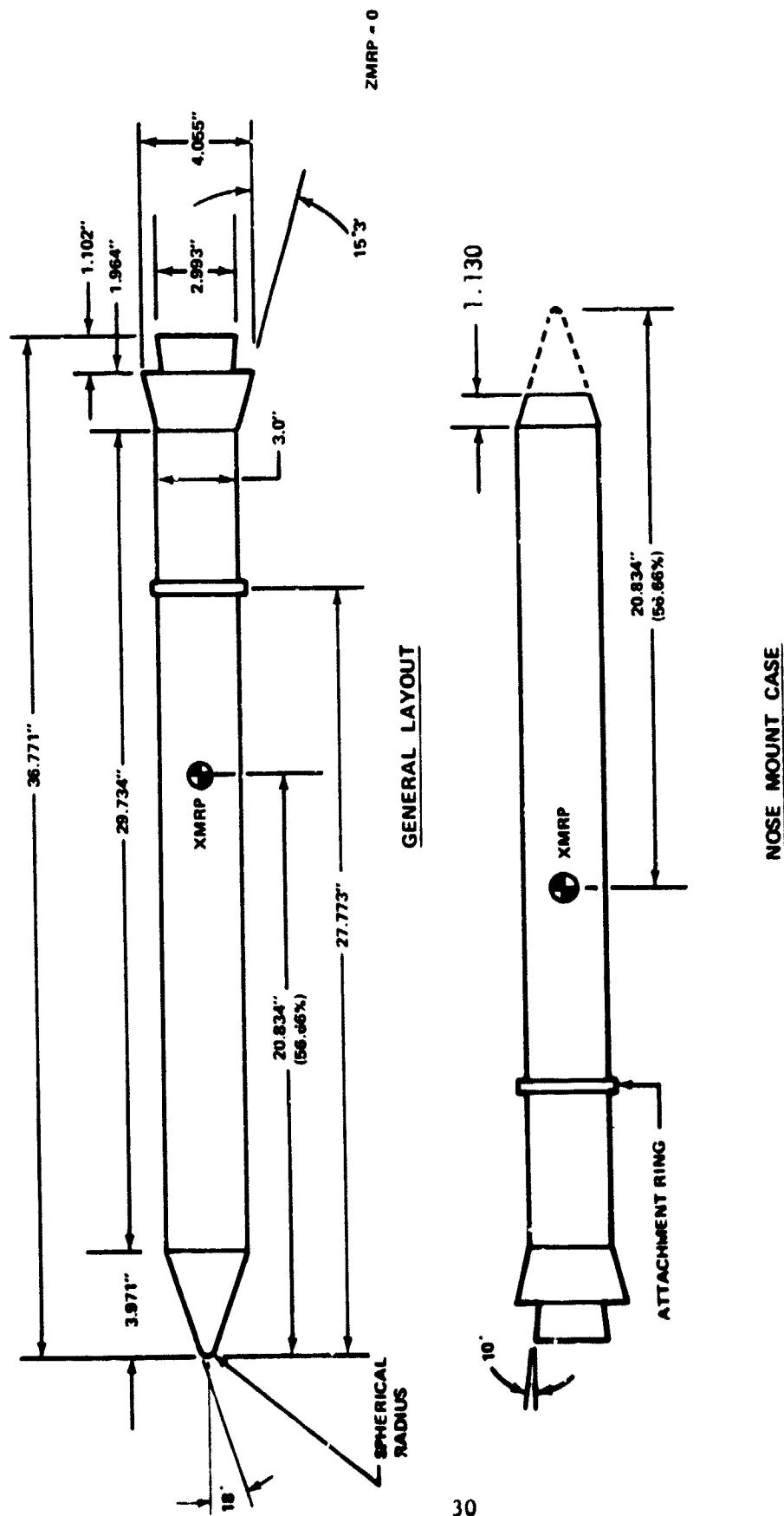


Figure 2. 142 INCH DIAMETER SRB - 0.02112 SCALE MODEL
(SRB CONFIGURATION 139 - MSFC MODEL #454)

(ENGINE SHROUD/NOZZLE SYMMETRIC WITH BODY)

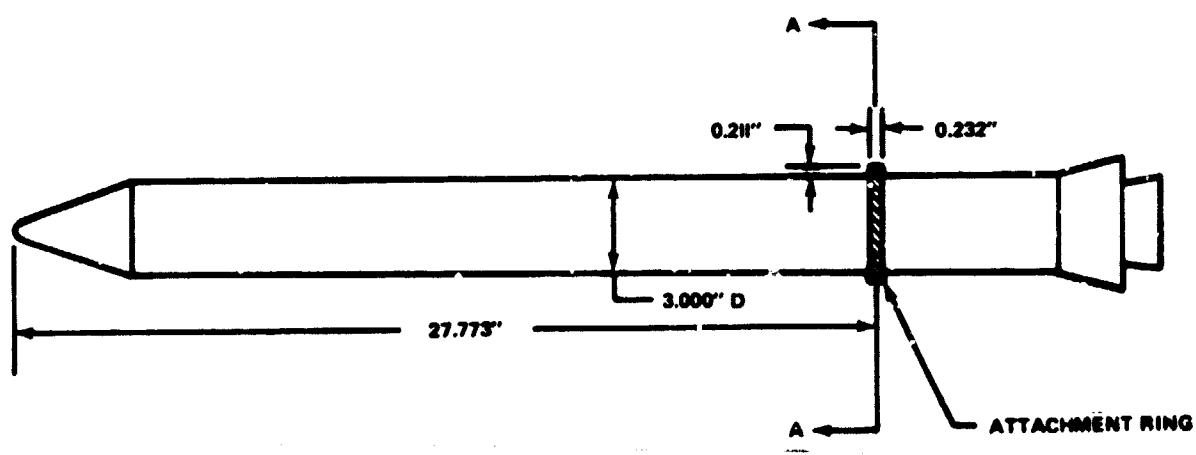
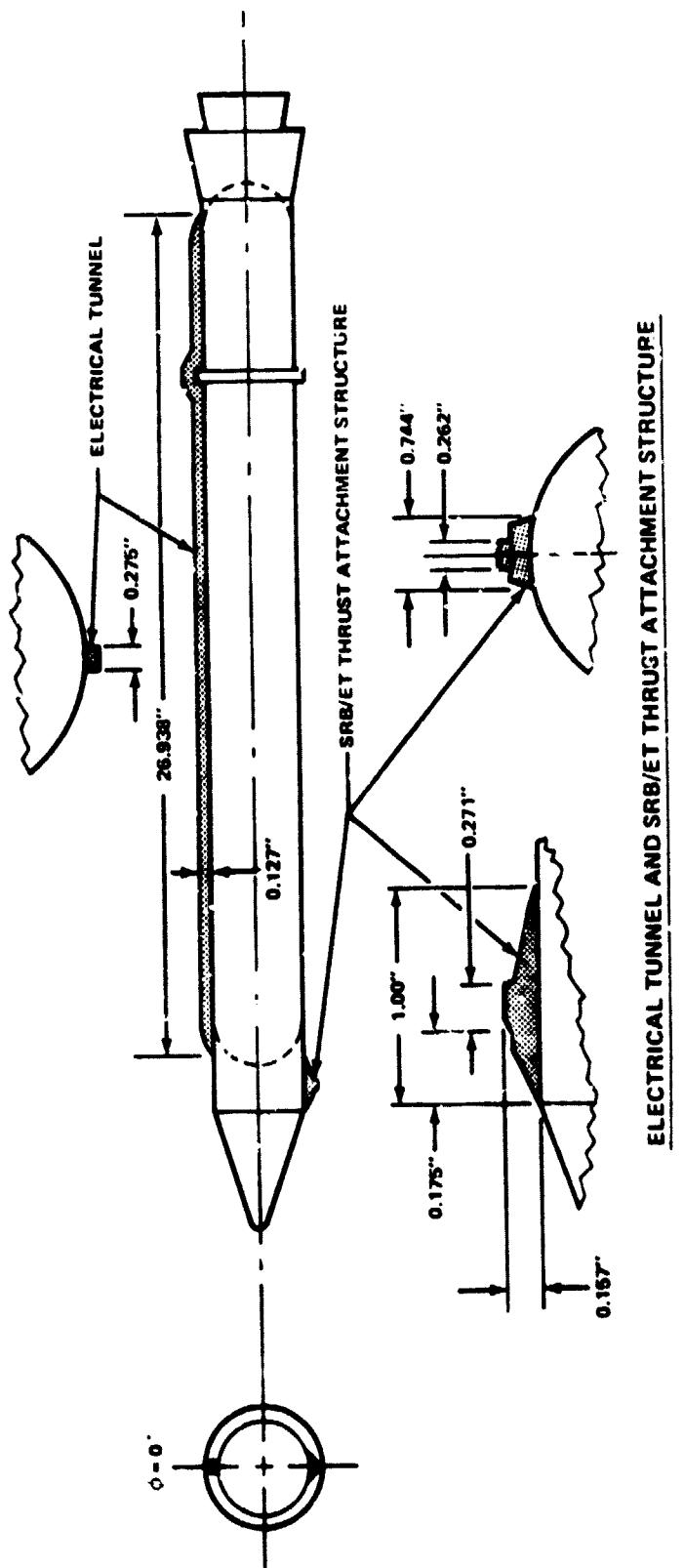


Figure 3. SRB/ET ATTACHMENT RING (MSFC MODEL #454)



NOTE: For the test at range, $\phi=0^\circ$ placed the thrust attachment structure on the windward side of the model and the electrical tunnel on the lee side of the model, both in the vertical pitch plane of the tunnel.

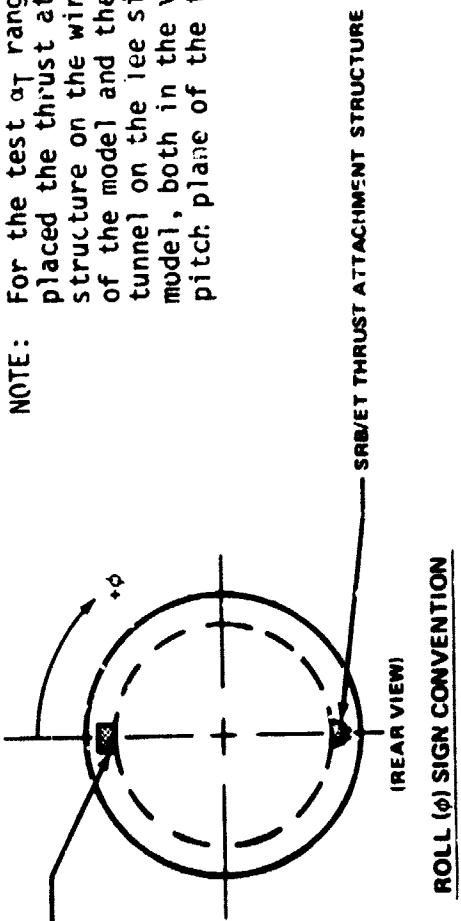


Figure 4. ELECTRICAL TUNNEL AND SRB/ET THRUST ATTACHMENT STRUCTURE (MSFC MODEL #454)

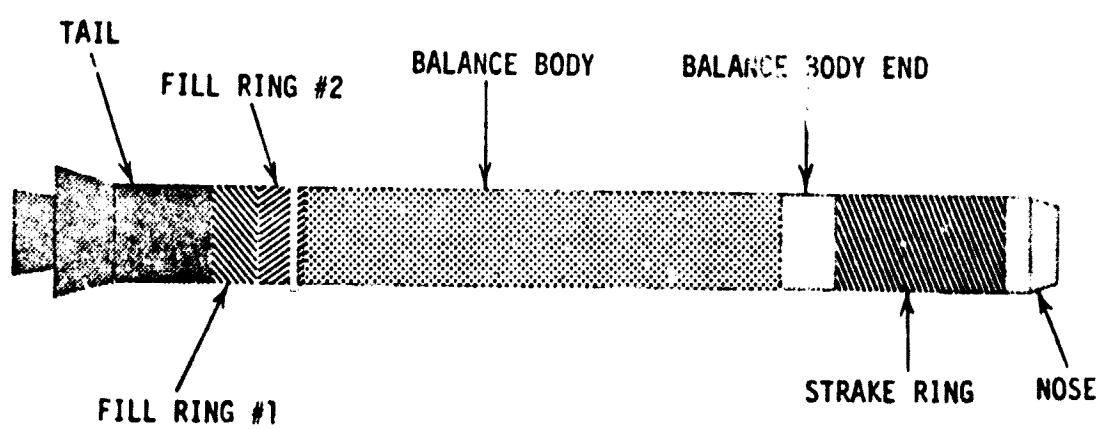


Figure 5. MAJOR MODEL SECTIONS

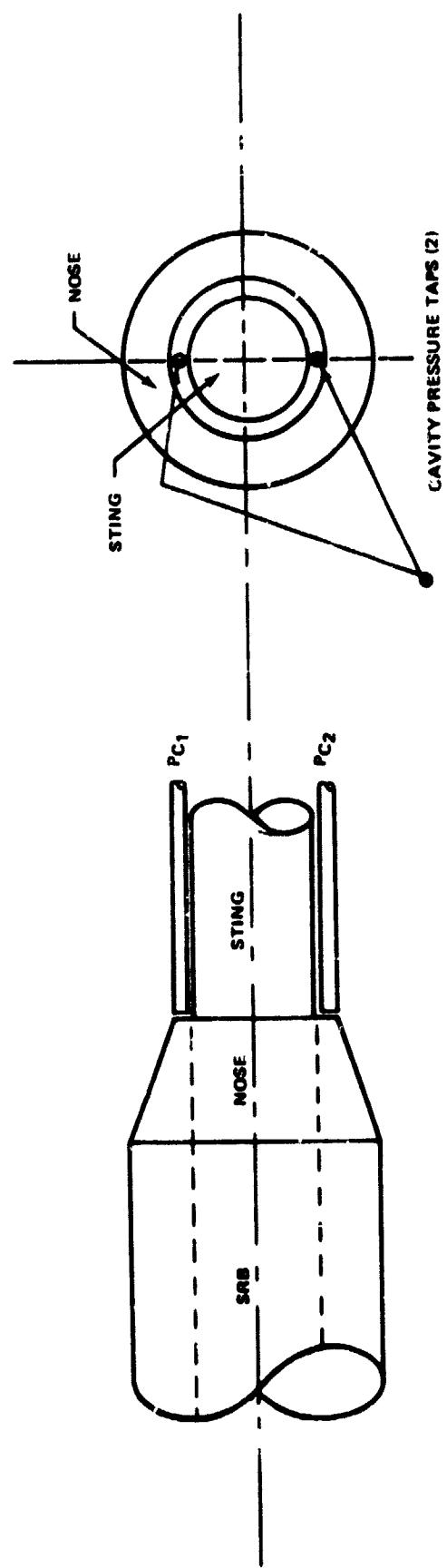


Figure 6. CAVITY PRESSURE TAP LOCATIONS

DATA FIGURES

SA-2SF LARC UPWT 1087 MSFC 454

(RH9001)

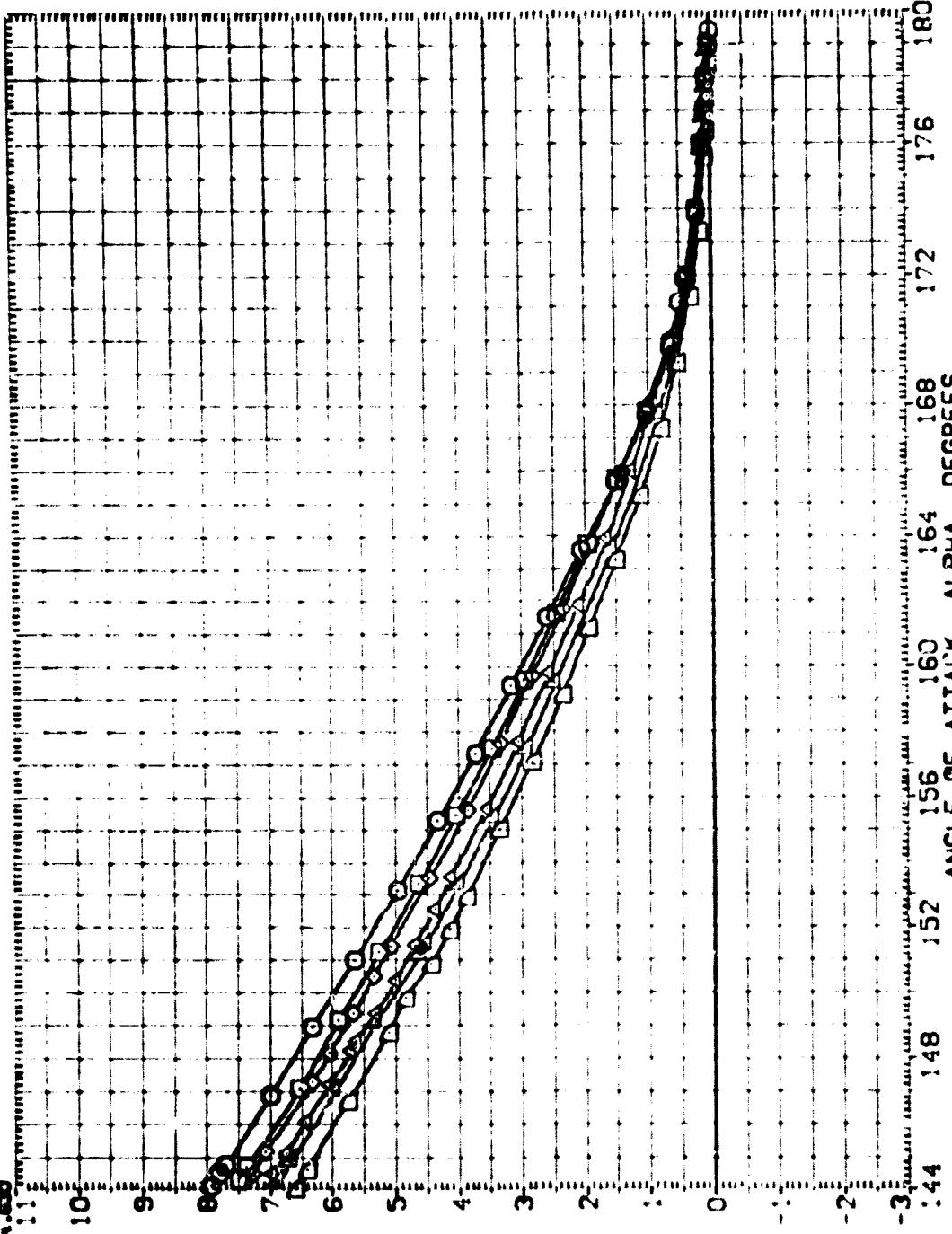
PARAMETRIC VALUES

| | | | |
|------------------|------|------|-------|
| NUO ₁ | .000 | PHI | .000 |
| NUO ₂ | .000 | BETA | 3.000 |
| NUO ₃ | .000 | A | 0 |
| NUO ₄ | .000 | C | 0 |
| NUO ₅ | .000 | E | 0 |
| NUO ₆ | .000 | | |

DYDODDO

REFERENCE INFORMATION

| | |
|-------|----------------|
| SPEF | 7.0000 INCHES |
| LREF | 3.0000 INCHES |
| BREF | 3.0000 INCHES |
| XTRP | 20.8340 INCHES |
| YTRP | .0000 INCHES |
| SCALE | .2211 |



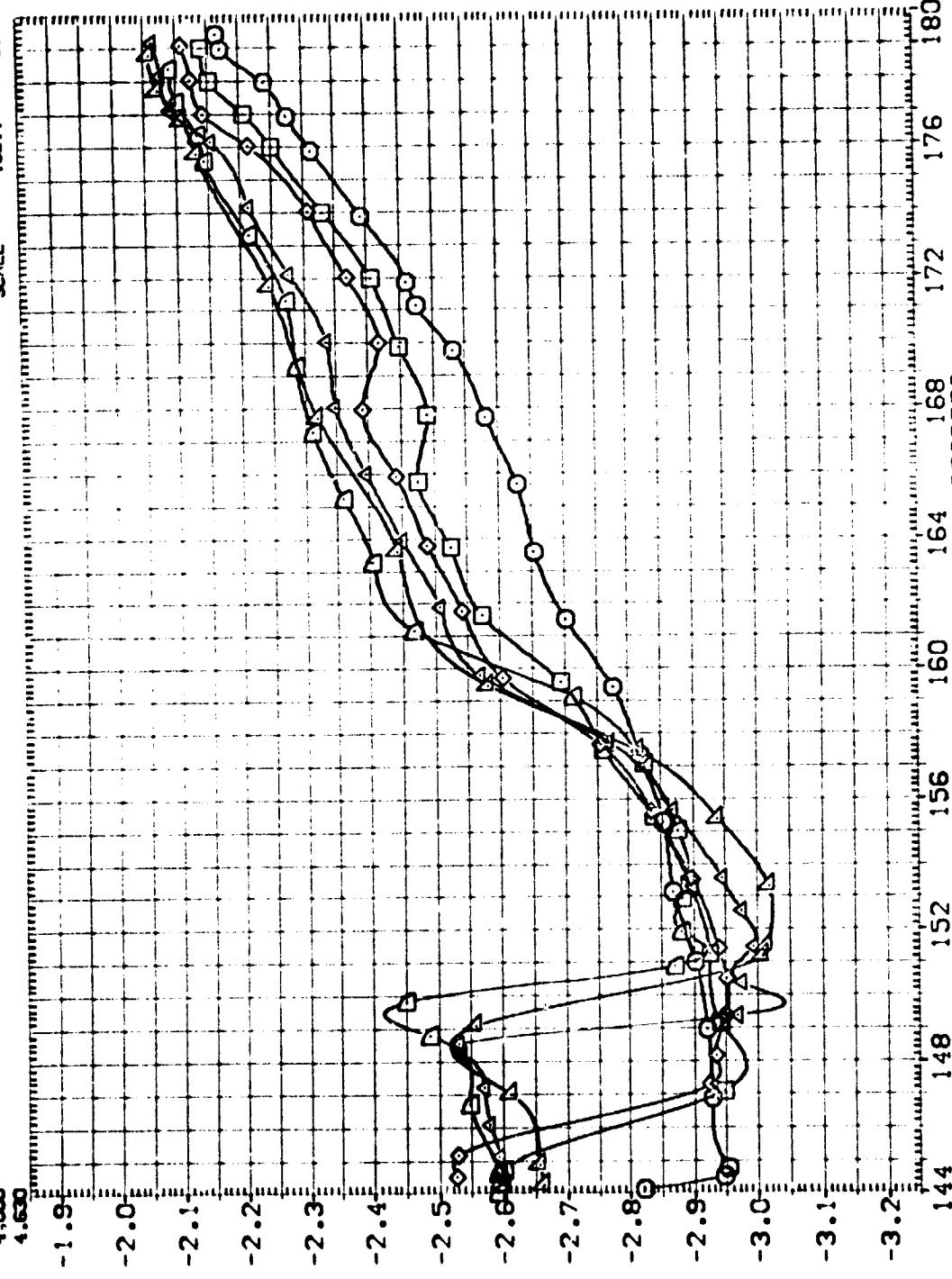
STATIC STABILITY CHARACTERISTICS OF SRB WITH ATTACHMENT RING

PAGE 1

SA-2SF LARC UPWT 1087 MSFC 454

(CRH9001)

| MACH | PARAMETRIC VALUES | Φ_{HI} |
|-------|-------------------|-------------|
| 2.300 | BETA .000 | .000 |
| 2.700 | A 3.000 | 1.000 |
| 2.900 | C .000 | 6.000 |
| 3.400 | E .000 | |
| 4.000 | | |
| 4.600 | | |



STATIC STABILITY CHARACTERISTICS OF SRB WITH ATTACHMENT RING

PAGE 2

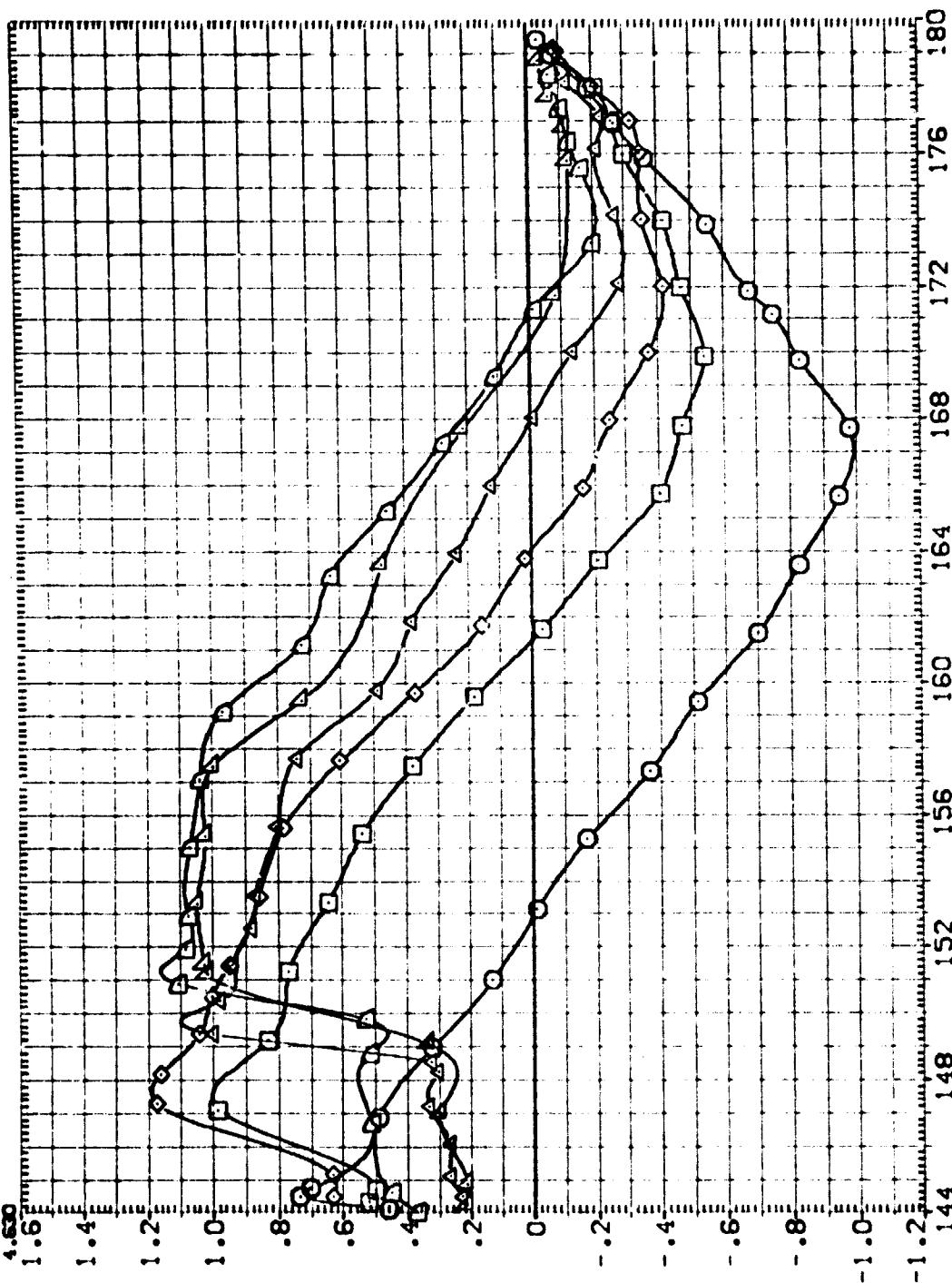
SA-25F LARC UPWT 1087 MSFC 454

MACH PARAMETRIC VALUES
2.300 .000 PH1
2.700 3.000 8
2.500 .000 0
3.400 .000 6.000
4.000 .000

REFERENCE INFORMATION
SREF 7.0850 50. IN.
LREF 3.0000 INCHES
BREF 3.0000 INCHES
XMP 20.8940 INCHES
YMP .0000 INCHES
ZMP .0211 SCALE

(RH9001)

MISSILE AXIS PITCHING MOMENT COEFFICIENT. CL/M



STATIC STABILITY CHARACTERISTICS OF SRB WITH ATTACHMENT RING

SA-25F LARC UPWT 1087 MSFC 454

(RH9001)

PARAMETRIC VALUES

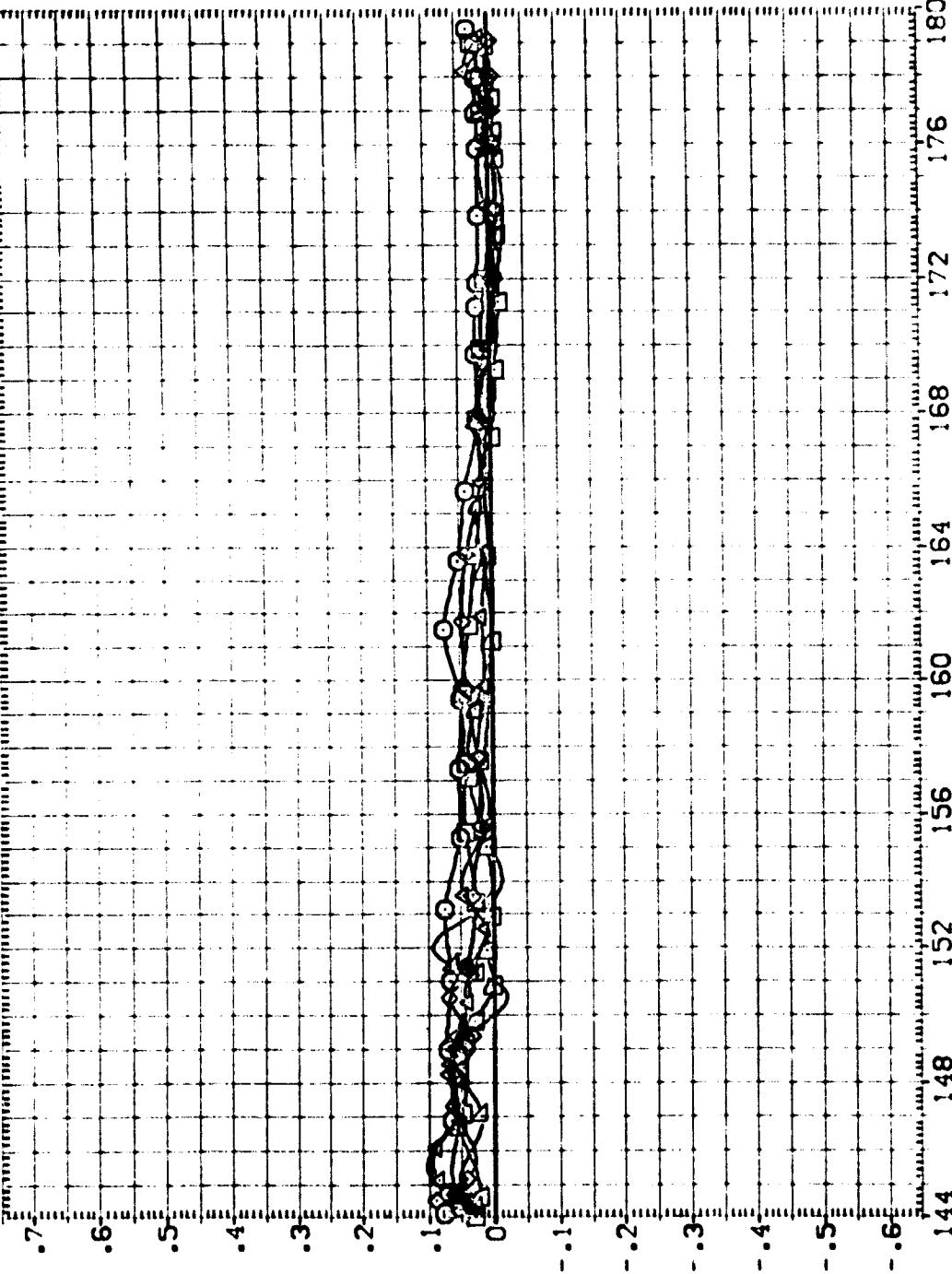
| | | | |
|-------|------|-------|-------|
| MACH | .000 | PHI | .000 |
| 2.300 | BETA | 3.000 | 9 |
| 2.700 | A | .000 | 1.000 |
| 2.950 | C | 0 | 6.000 |
| 3.450 | E | .000 | |
| 4.000 | | | |
| 4.650 | | | |

REFERENCE INFORMATION

| | | | |
|-------|---------|-----------|--------|
| SREF | 7.0680 | SCAL. IN. | .00003 |
| LREF | 3.0000 | INCHES | |
| XMRP | 3.0000 | INCHES | |
| YMRP | 20.8340 | INCHES | |
| ZMRP | .0000 | INCHES | |
| SCALE | .0311 | | |

SYMBOLS
O □ ◇ △ ▲ ▽ ▵ ▷

MISSILE AXIS SIDE FORCE COEFFICIENT. CYM



STATIC STABILITY CHARACTERISTICS OF SRB WITH ATTACHMENT RING

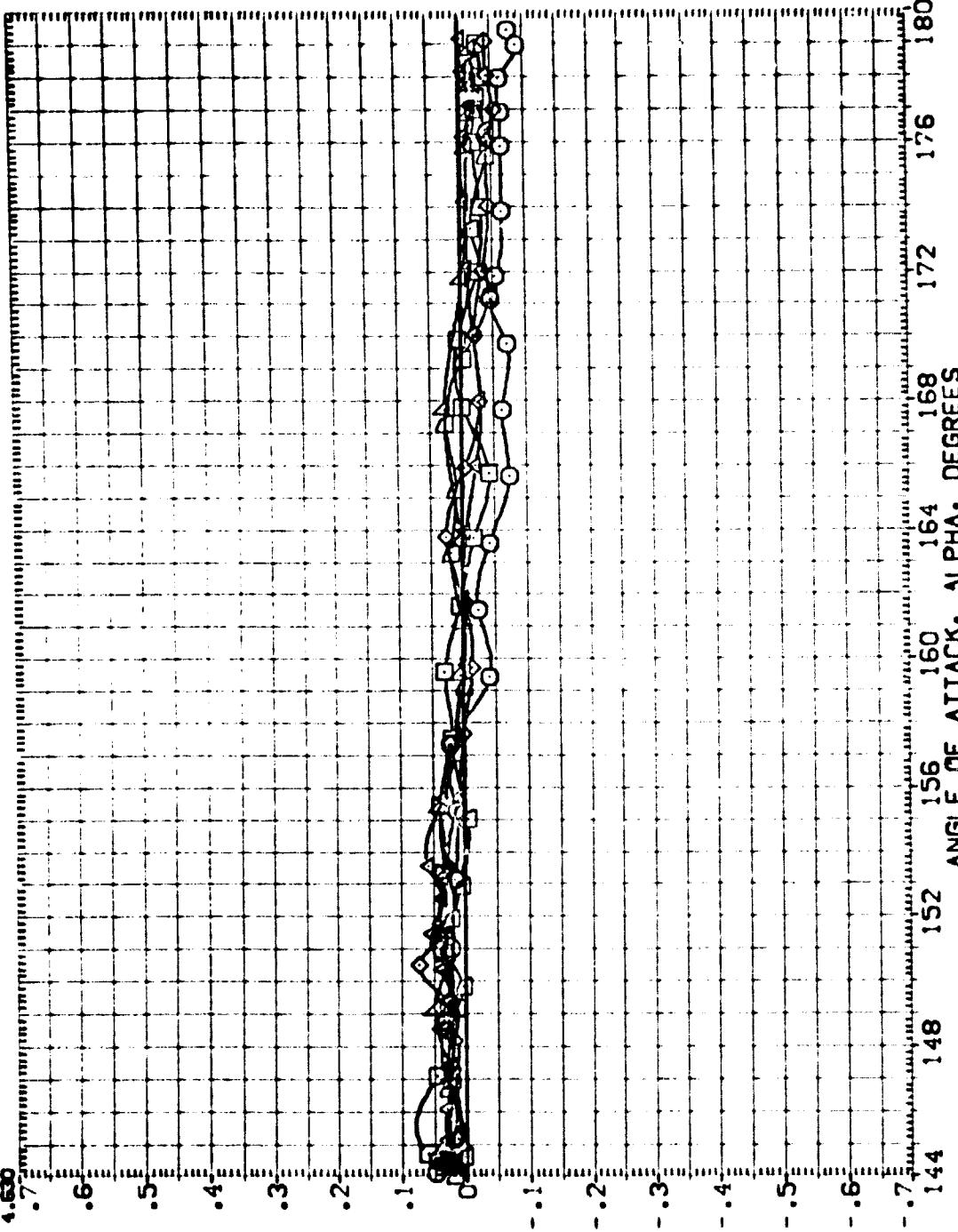
PAGE 4

SA-25F LARC UPWT 1087 MSFC 454

(RH9001)



REFERENCE INFORMATION
SREF 7.0880 SQ. IN.
LREF 3.0000 NOES
BREF 3.0000 NOES
XMRP 20.9310 NOES
YMRP .0000 NOES
ZMRP .0211 SCALE



MISSILE AXIS YAWING MOMENT COEFFICIENT, CYNM

STATIC STABILITY CHARACTERISTICS OF SRB WITH ATTACHMENT RING

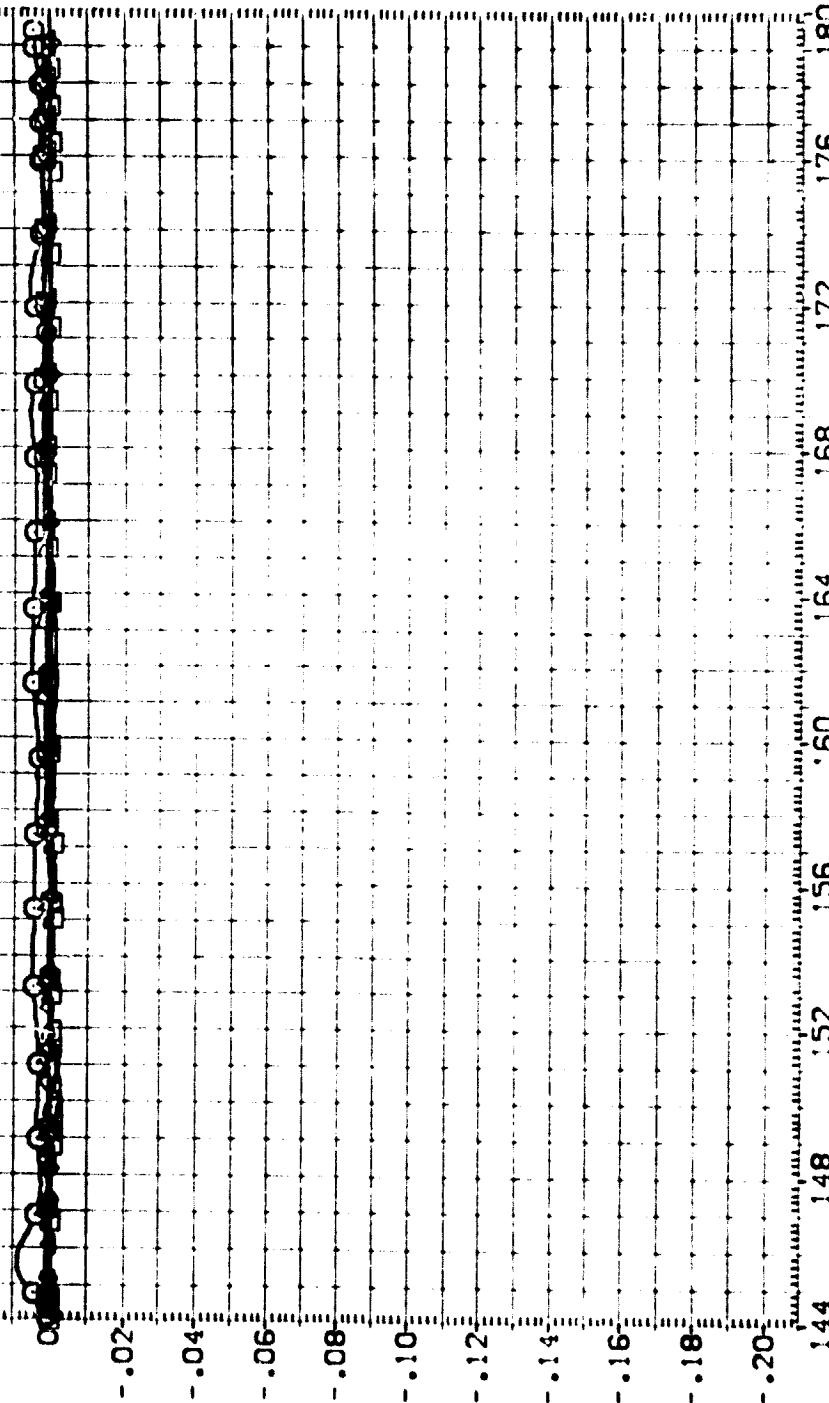
SA-25F LARC UPWT 1087 MSFC 454

(CRH9001)

| MACH | PARAMETRIC VALUES |
|------|--------------------|
| 2.30 | BETA .000 PHI .000 |
| 2.70 | A 3.000 B 1.000 |
| 2.90 | C .000 D 6.000 |
| 3.40 | E .000 |
| 4.00 | |
| 4.60 | |

REFERENCE INFORMATION
SREF 7.0880 SP. IN.
LREF 3.0000 INCHES
BREF 3.0000 INCHES
XMRD 20.8340 INCHES
YMRP .0000 INCHES
ZMRP .0211 SCALE

MISSILE AXIS ROLLING MOMENT COEFFICIENT. CBL

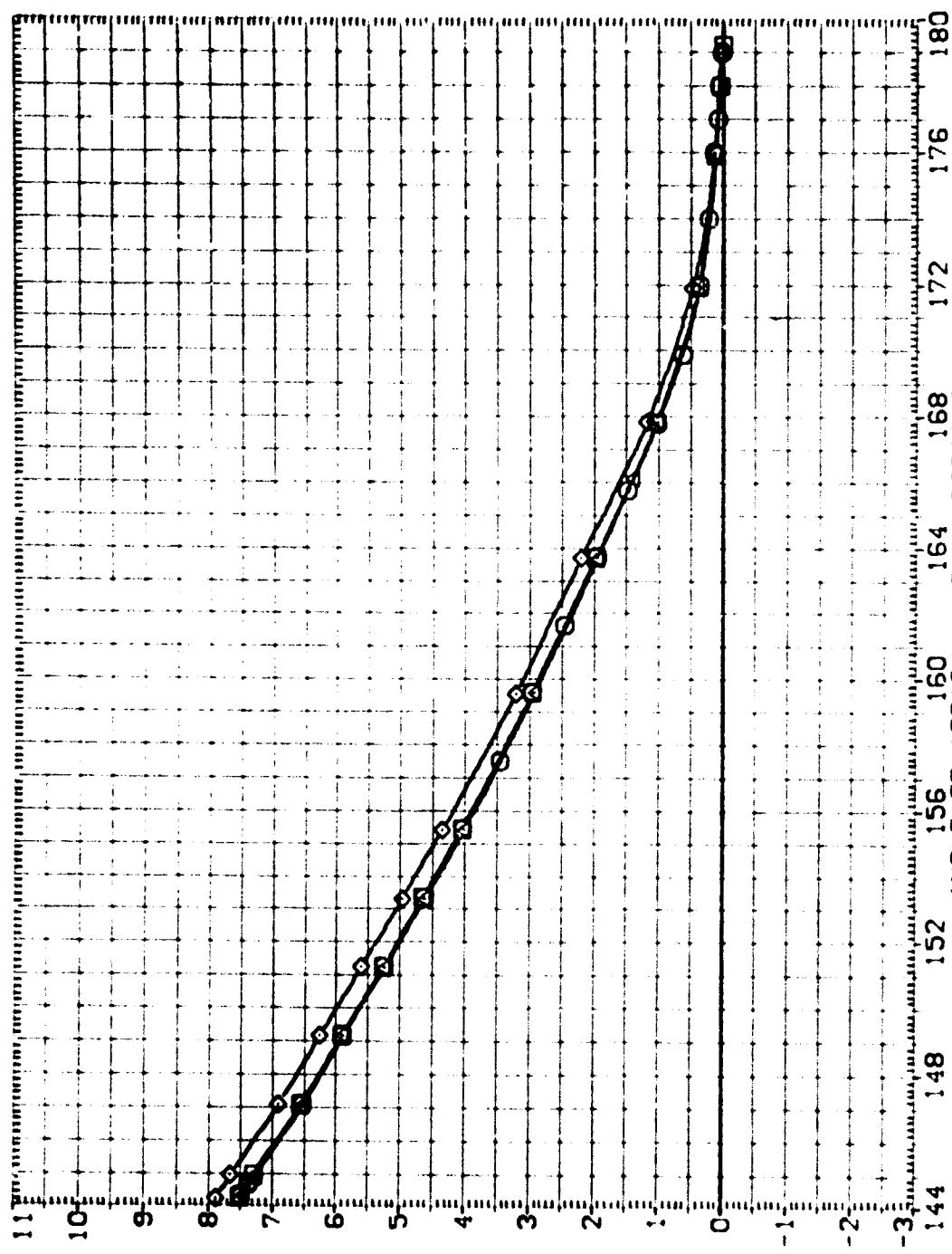


STATIC STABILITY CHARACTERISTICS OF SRB WITH ATTACHMENT RING
144 148 :52 156 160 164 168 172 176 180
ANGLE OF ATTACK. ALPHA. DEGREES

DATA SET NO. 1 CONVENTION DESCRIPTION

| | | | | | | |
|--------|---|-------------|------------------|---------------|------|------|
| A-5001 | D | SA-25° LARC | YAW 108° MFC 154 | ELT TH ATTACH | .000 | .000 |
| A-5002 | D | SA-25° LARC | YAW 108° MFC 154 | ELT TH ATTACH | .45 | .000 |
| A-5003 | X | SA-25° LARC | YAW 108° MFC 154 | ELT TH ATTACH | .50 | .000 |
| A-5004 | X | SA-25° LARC | YAW 108° MFC 154 | ELT TH ATTACH | .135 | .000 |

REFERENCE INFORMATION
 SREF 7.0690 SQ. IN.
 LREF 3.0000 INCHES
 BREF .0000 INCHES
 XHYP 20.8310 INCHES
 YHYP .0000 INCHES
 ZHYP .3211 SCALE



MISSILE AXIS NORMAL FORCE COEFFICIENT, CNM

EFFECT OF ROLL ANGLE ON STATIC STABILITY CHARACTERISTICS OF SRB
 $(\text{C}_M \text{MACH} = 2.70)$

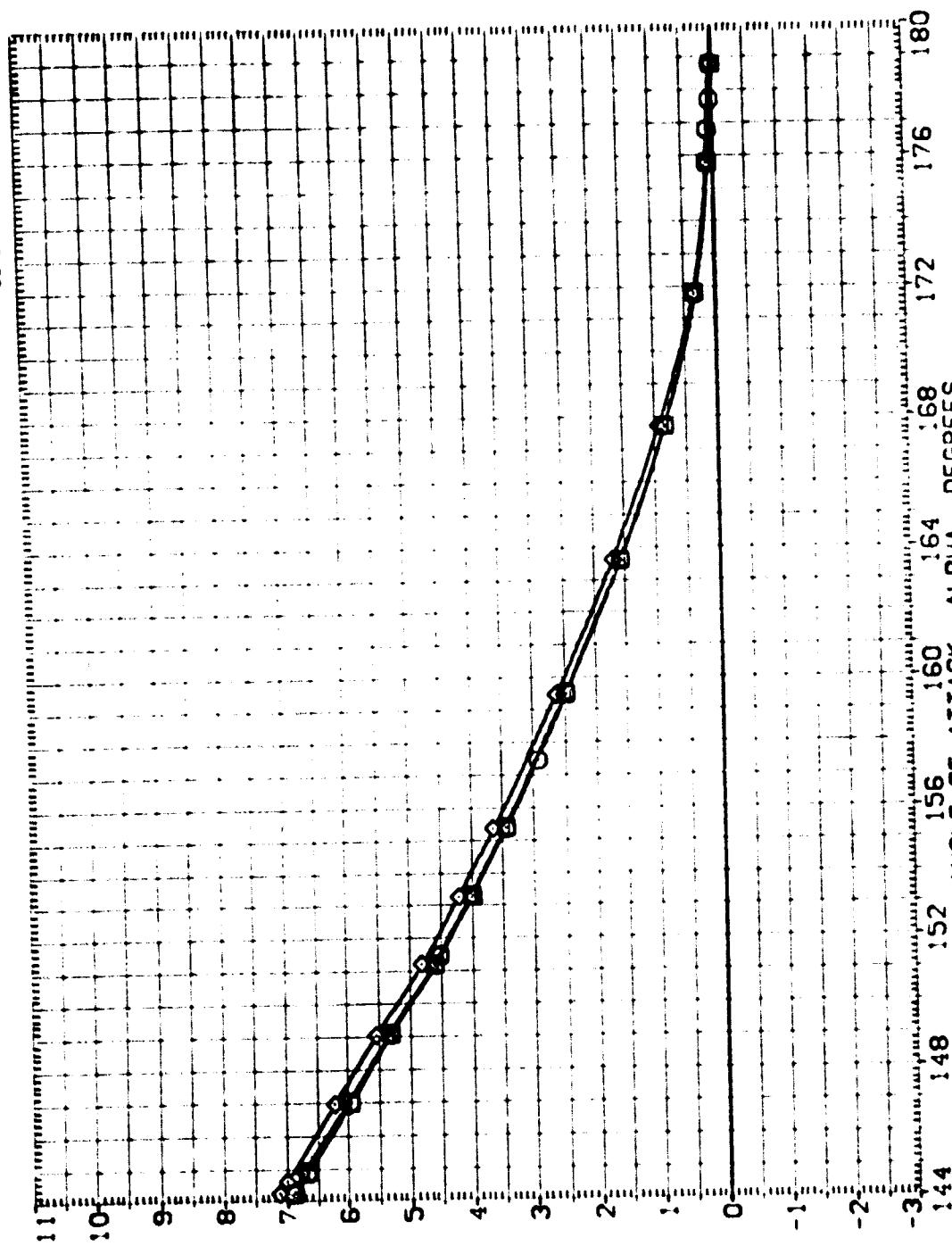
DATA SET SYMBOL CONFIGURATION DESCRIPTION

| | | | | | |
|---|------------------|----------|-----|-----|-----------|
| Q | 50-25° LARE SPNT | 107° HSC | 45° | ELT | TH ATTACH |
| △ | 50-25° LARE SPNT | 107° HSC | 45° | ELT | TH ATTACH |
| ○ | 50-25° LARE SPNT | 107° HSC | 45° | ELT | TH ATTACH |
| × | 50-25° LARE SPNT | 107° HSC | 45° | ELT | TH ATTACH |
| □ | 50-25° LARE SPNT | 107° HSC | 45° | ELT | TH ATTACH |

SCALE

REFERENCE INFORMATION

| | | |
|------|---------|---------|
| SREF | 7.0830 | SD. IN. |
| LREF | 3.0000 | INCHES |
| BREF | 3.0000 | INCHES |
| XHPP | 20.8310 | INCHES |
| YHPP | .0000 | INCHES |
| ZHPP | .0211 | SCALE |



EFFECT OF ROLL ANGLE ON STATIC STABILITY CHARACTERISTICS OF SRB

(B)MACH = 4.00

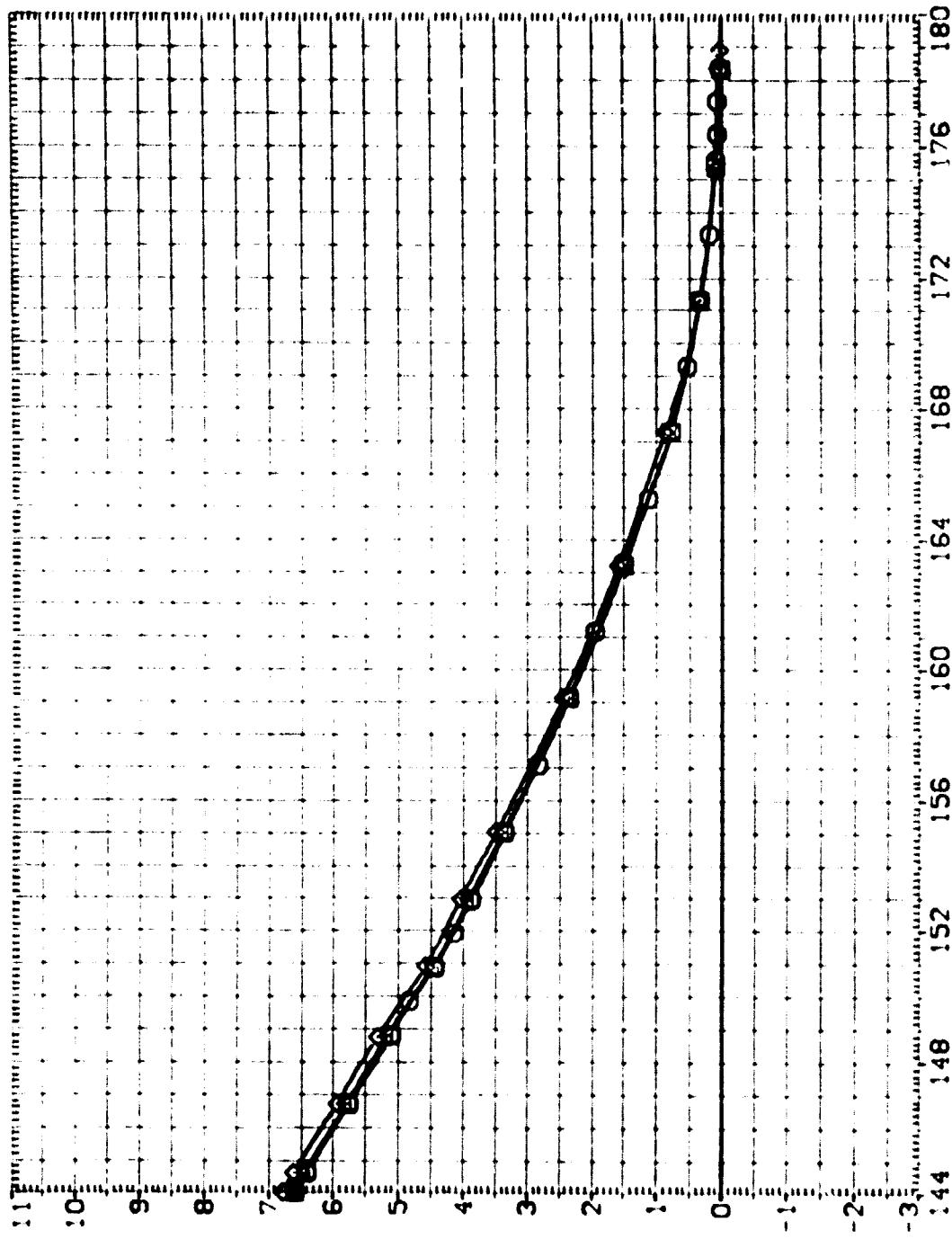
PAGE 8

DATA SET SYMBOL CONFIGURATION DESCRIPTION

| | | |
|----------|---|------------------------------|
| [A-9001] | □ | SA-25 LARC UNIT 1087 MFC 454 |
| [A-9002] | ○ | SA-25 LARC UNIT 1087 MFC 454 |
| [A-9003] | × | SA-25 LARC UNIT 1087 MFC 454 |
| [A-9004] | △ | SA-25 LARC UNIT 1087 MFC 454 |

REFERENCE INFORMATION

| | | |
|------|---------|---------|
| SREF | 7.0850 | SD. IN. |
| LREF | 3.0000 | INCHES |
| BREF | 3.0000 | INCHES |
| XMAP | 20.8340 | INCHES |
| YMAP | .0000 | INCHES |
| ZMAP | .0211 | SCALE |



MISSILE AXIS NORMAL FORCE COEFFICIENT. CNM

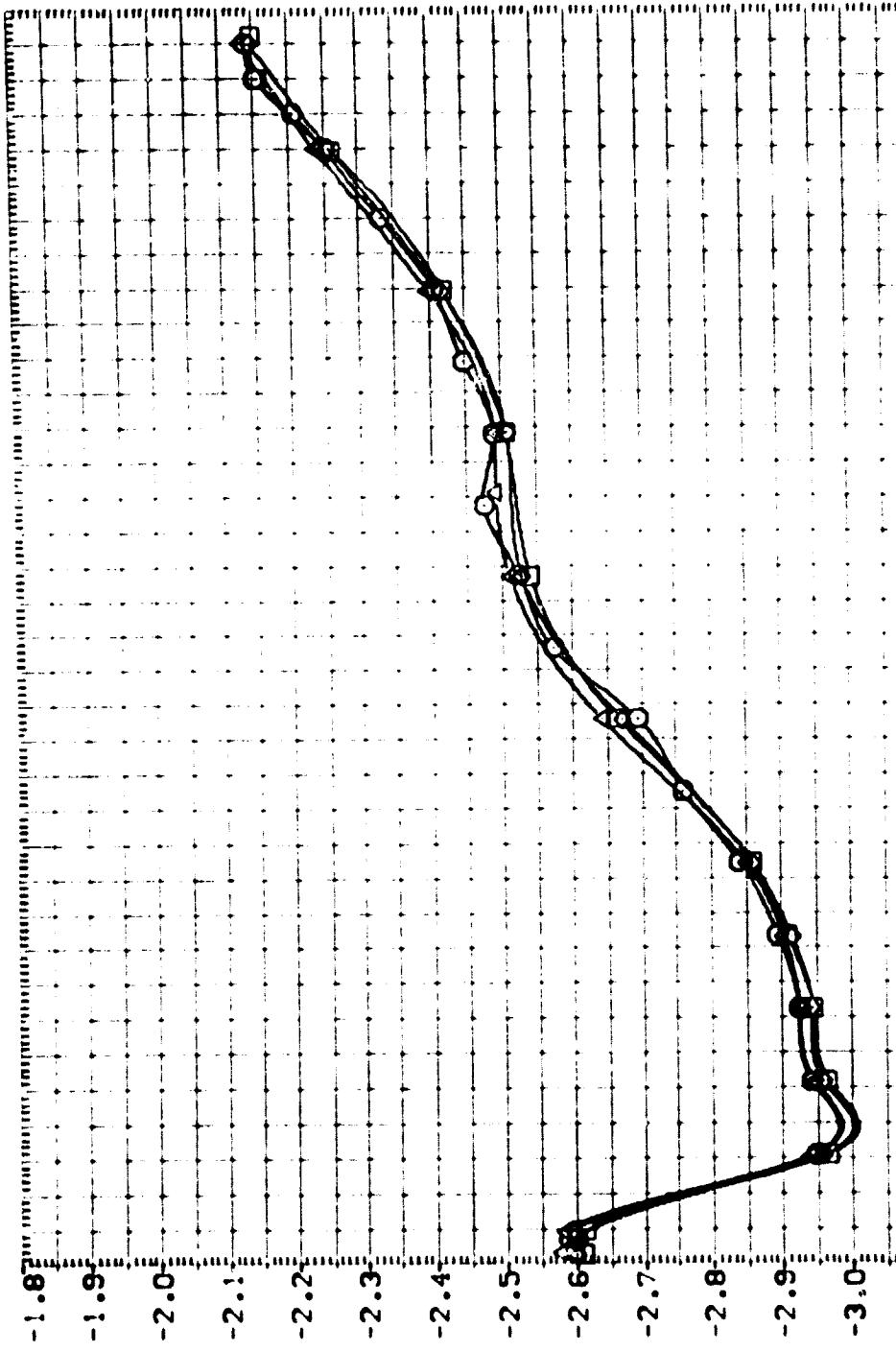
EFFECT OF ROLL ANGLE ON STATIC STABILITY CHARACTERISTICS OF SRB

(C)_{MACH} = 4.63

DATA SET SYMBOL CONFIGURATION DESCRIPTION

| | | | |
|-------|-------|-----------|-------------|
| AMACH | SA-25 | LARE UNIT | 107 MFC 154 |
| AMACH | SA-25 | LARE UNIT | 107 MFC 154 |
| AMACH | SA-25 | LARE UNIT | 107 MFC 154 |
| AMACH | SA-25 | LARE UNIT | 107 MFC 45X |
| AMACH | SA-25 | LARE UNIT | 107 MFC 45X |

REFERENCE INFORMATION
SREF 7.0690 SQ IN.
LREF .0000 INCHES
BREF .0000 INCHES
XREF 20.8340 INCHES
YREF .0000 INCHES
ZREF .0211 SCALE



MISSILE AXIS AXIAL FORCE COEFFICIENT. CA

180
176
172
168
164
160
156
152
148
144

EFFECT OF ROLL ANGLE ON STATIC STABILITY CHARACTERISTICS OF SRB

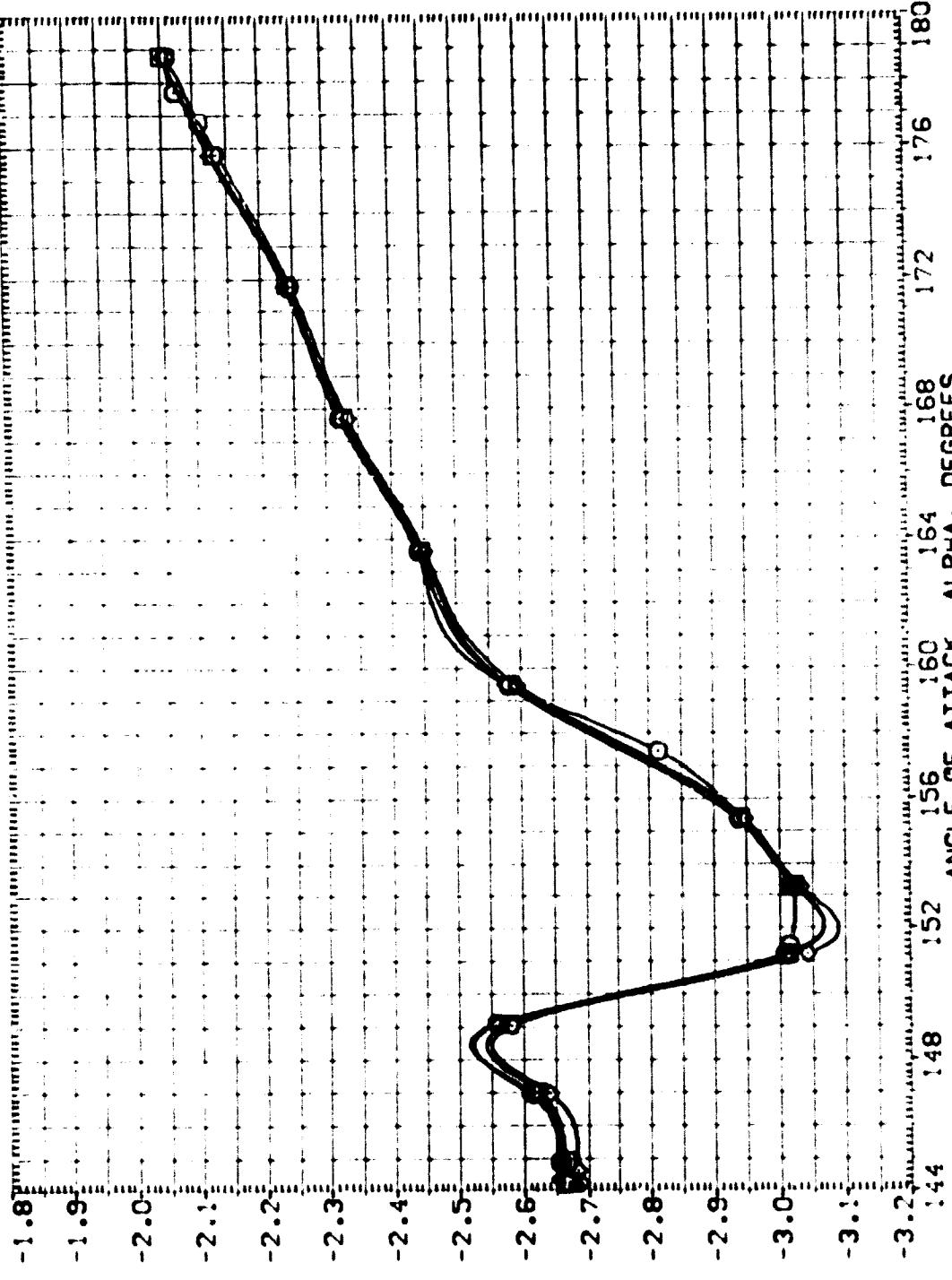
CA/MACH = 2.70

PAGE 10

DATA SET NUMBER CONVENTION DESCRIPTION

| | | | | | |
|--------|------------------|------|---------|-----|-----------|
| 100001 | SA-25F LARC SPAN | 1087 | MFC 454 | ELT | TH ATTACH |
| 100002 | SA-25F LARC SPAN | 1087 | MFC 454 | ELT | TH ATTACH |
| 100003 | SA-25F LARC SPAN | 1087 | MFC 454 | ELT | TH ATTACH |
| 100004 | SA-25F LARC SPAN | 1087 | MFC 454 | ELT | TH ATTACH |

REFERENCE INFORMATION
 SREF 7.0690 90. IN.
 LREF 3.0000 100ES
 BREF 3.0000 100ES
 XREF 20.8310 100ES
 YREF .0000 100ES
 ZREF .0000 100ES
 SCALE .0211



MISSILE AXIS AXIAL FORCE COEFFICIENT, CA

EFFECT OF ROLL ANGLE ON STATIC STABILITY CHARACTERISTICS OF SRB

(B)MACH = 4.00

PAGE 11

1

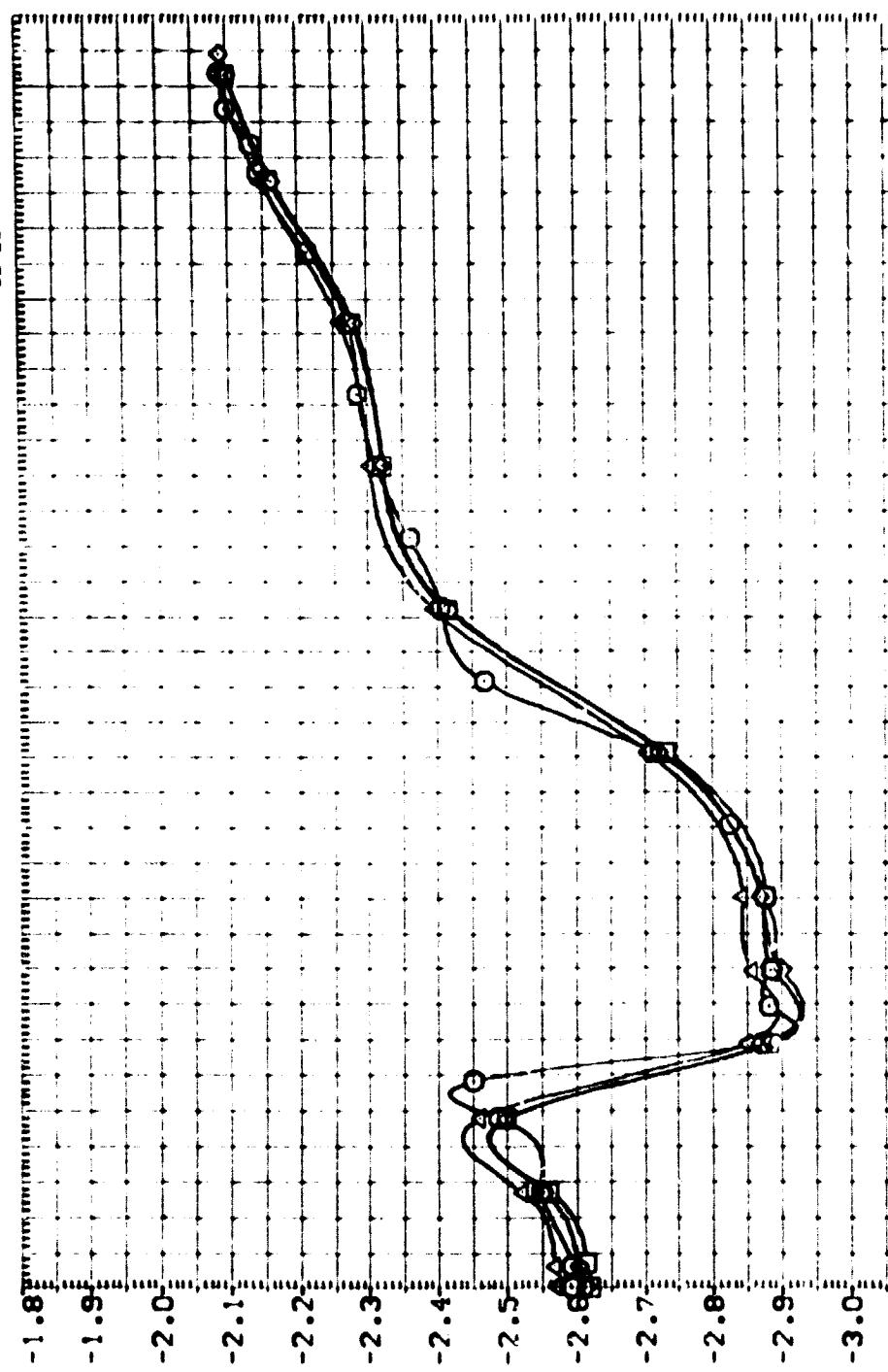
DATA SET SYMBOL CONFIGURATION DESCRIPTION

| | | | | |
|--------|---|------------------|------|----------|
| L40001 | O | SA-25F LARE UNIT | 1087 | HSFC 154 |
| L40002 | X | SA-25F LARE UNIT | 1087 | HSFC 154 |
| L40003 | + | SA-25F LARE UNIT | 1087 | HSFC 154 |
| L40004 | * | SA-25F LARE UNIT | 1087 | HSFC 154 |

REFERENCE INFORMATION

| | | |
|-------|---------|--------|
| SREF | .0680 | SOLES |
| LREF | 3.0000 | 1.90ES |
| BREF | 3.0000 | 1.90ES |
| XMP | 20.8310 | 1.90ES |
| YMP | .0000 | 1.90ES |
| ZMP | .0000 | 1.90ES |
| SCALE | .0211 | |

PHI BETA



MISSILE AXIS AXIAL FORCE COEFFICIENT. CA

1.44 1.48 1.52 1.56 1.60 1.64 1.68 1.72 1.76 1.80
-3.2 -3.1 -3.0 -2.9 -2.8 -2.7 -2.6 -2.5 -2.4 -2.3 -2.2 -2.1 -2.0 -1.9 -1.8

EFFECT OF ROLL ANGLE ON STATIC STABILITY CHARACTERISTICS OF SRB

(C)_{MACH} = 4.63

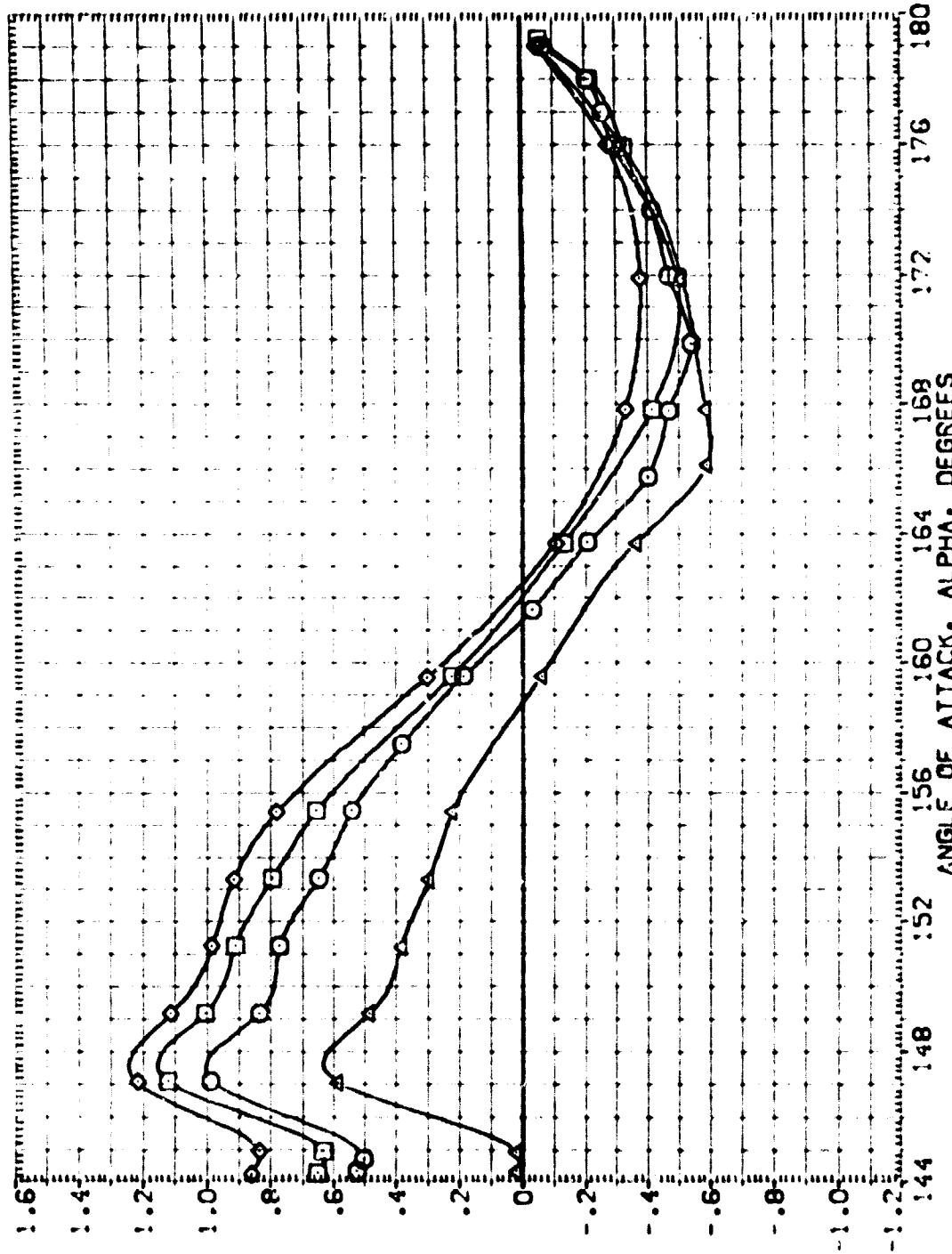
PAGE 12

DATA SET 7 static configuration description

| | |
|--------|-----------|
| ANGLE | 144 |
| LARC | SU |
| ROLL | 1087 |
| YAW | 154 |
| PITCH | 45 |
| ELT | TH ATTACH |
| REF | 90 |
| XHPP | 000 |
| YHPP | 000 |
| ZHPP | 000 |
| ATTACH | 135 |
| ELT | TH ATTACH |
| REF | 000 |
| XHPP | 000 |
| YHPP | 000 |
| ZHPP | 000 |
| ATTACH | 135 |
| ELT | TH ATTACH |
| REF | 000 |
| XHPP | 000 |
| YHPP | 000 |
| ZHPP | 000 |
| ATTACH | 135 |
| ELT | TH ATTACH |
| REF | 000 |
| XHPP | 000 |
| YHPP | 000 |
| ZHPP | 000 |
| ATTACH | 135 |

REFERENCE INFORMATION

| | |
|-------|---------|
| SREF | 7.080 |
| LREF | .000 |
| BREF | 3.000 |
| CREF | 3.000 |
| XHPP | 20.8040 |
| YHPP | .0000 |
| ZHPP | .0000 |
| SCALE | .3211 |



EFFECT OF ROLL ANGLE ON STATIC STABILITY CHARACTERISTICS OF SRB

MASS = 2.70

PAGE 13

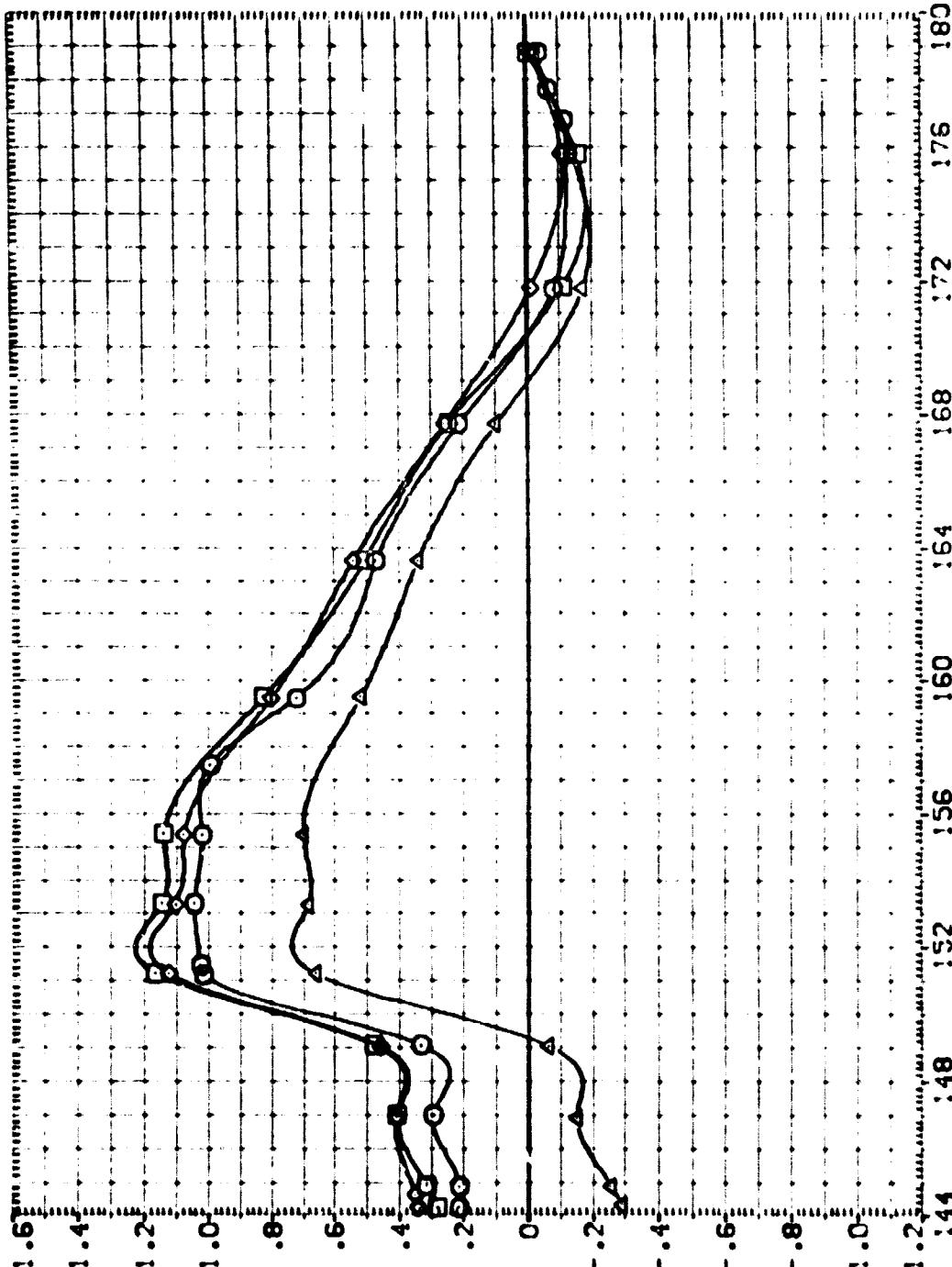
EFFECT OF ROLL ANGLE ON STATIC STABILITY CHARACTERISTICS OF SRB

MACH = 4.00

PAGE 14

| REFERENCE INFORMATION | |
|-----------------------|--------|
| REF | 7.0000 |
| LREF | .0000 |
| BREF | .0000 |
| XREF | .0000 |
| YREF | .0000 |
| ZREF | .0000 |
| SCALE | .0011 |

| DATA SET NUMBER | CONFIGURATION DESCRIPTION | ROLL ANGLE | ATTACK ANGLE | PITCH ANGLE | ROLL COEF. | ATTACK COEF. | PITCH COEF. |
|-----------------|---------------------------|------------|--------------|-------------|------------|--------------|-------------|
| 1 | 1 | 15.0 | 15.0 | 15.0 | 1.12 | 0.96 | 0.92 |
| 2 | 2 | 30.0 | 15.0 | 15.0 | 1.10 | 0.95 | 0.90 |
| 3 | 3 | 45.0 | 15.0 | 15.0 | 1.05 | 0.94 | 0.88 |
| 4 | 4 | 60.0 | 15.0 | 15.0 | 1.00 | 0.92 | 0.85 |
| 5 | 5 | 75.0 | 15.0 | 15.0 | 0.95 | 0.88 | 0.80 |
| 6 | 6 | 90.0 | 15.0 | 15.0 | 0.85 | 0.80 | 0.70 |
| 7 | 7 | 105.0 | 15.0 | 15.0 | 0.75 | 0.70 | 0.60 |
| 8 | 8 | 120.0 | 15.0 | 15.0 | 0.65 | 0.60 | 0.50 |
| 9 | 9 | 135.0 | 15.0 | 15.0 | 0.55 | 0.50 | 0.40 |
| 10 | 10 | 150.0 | 15.0 | 15.0 | 0.45 | 0.40 | 0.30 |
| 11 | 11 | 165.0 | 15.0 | 15.0 | 0.35 | 0.30 | 0.20 |
| 12 | 12 | 180.0 | 15.0 | 15.0 | 0.25 | 0.20 | 0.10 |

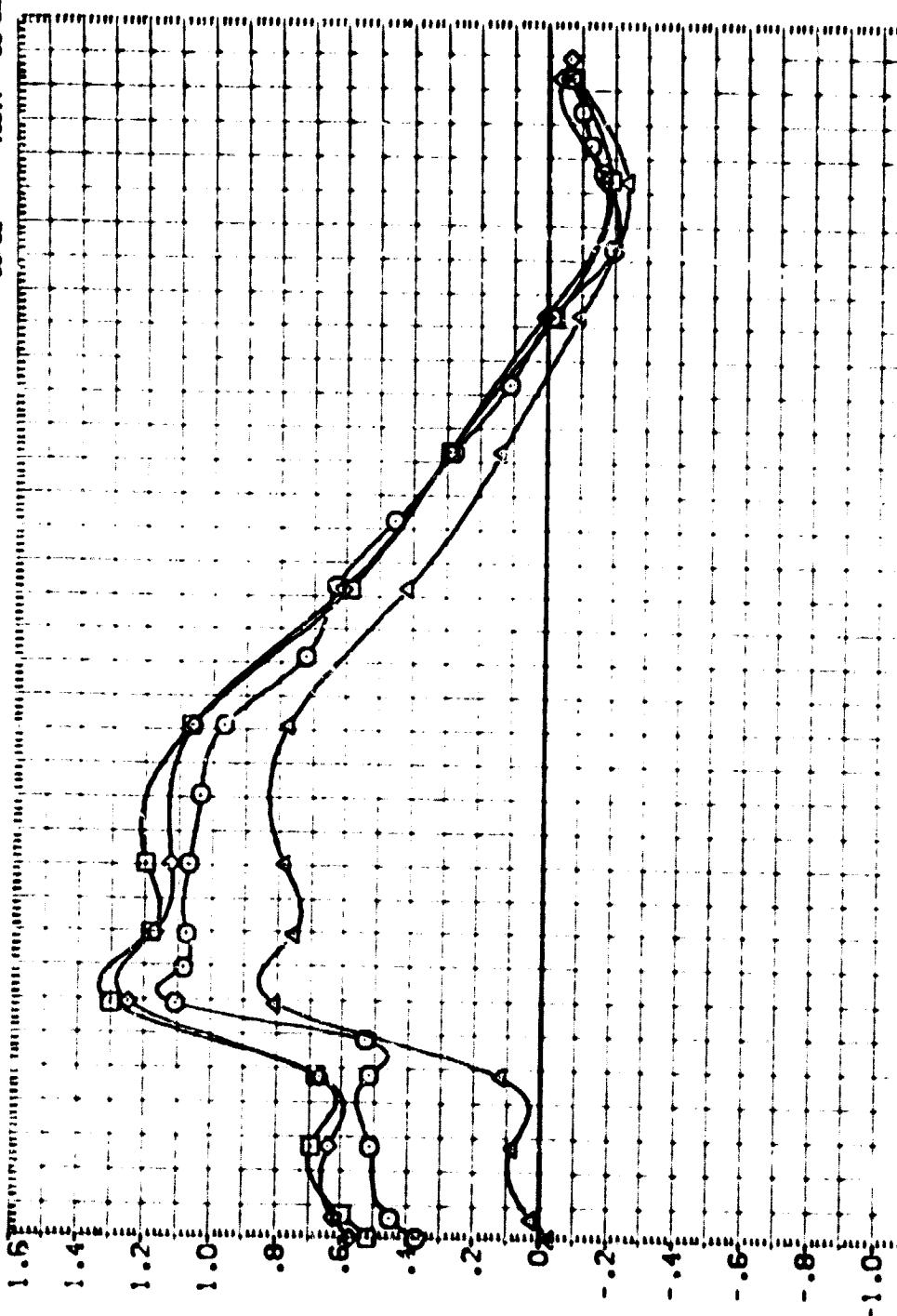


MISSILE AXIS PITCHING MOMENT COEFFICIENT, CLM

DATA SET 1
STABILITY INFORMATION DESCRIPTION

| | | | | | | |
|----------------|----------------|----------------|----------------|----------------|----------------|----------------|
| 144 | 148 | 152 | 156 | 160 | 164 | 168 |
| 30-25 | 30-25 | 30-25 | 30-25 | 30-25 | 30-25 | 30-25 |
| LAC |
| 100% ATTACH |
| MFC 45° |
| ELT |
| TW ATTACH |
| 135.000 | 135.000 | 135.000 | 135.000 | 135.000 | 135.000 | 135.000 |

REFERENCE INFORMATION
 SREF 7.0680 SO IN.
 LREF 3.0000 200ES
 GREF 3.0000 200ES
 XHDP 20.8310 199ES
 YHDP .0000 199ES
 ZHDP .2211 199ES
 SCALE



MISSILE AXIS PITCHING MOMENT COEFFICIENT, CLMM

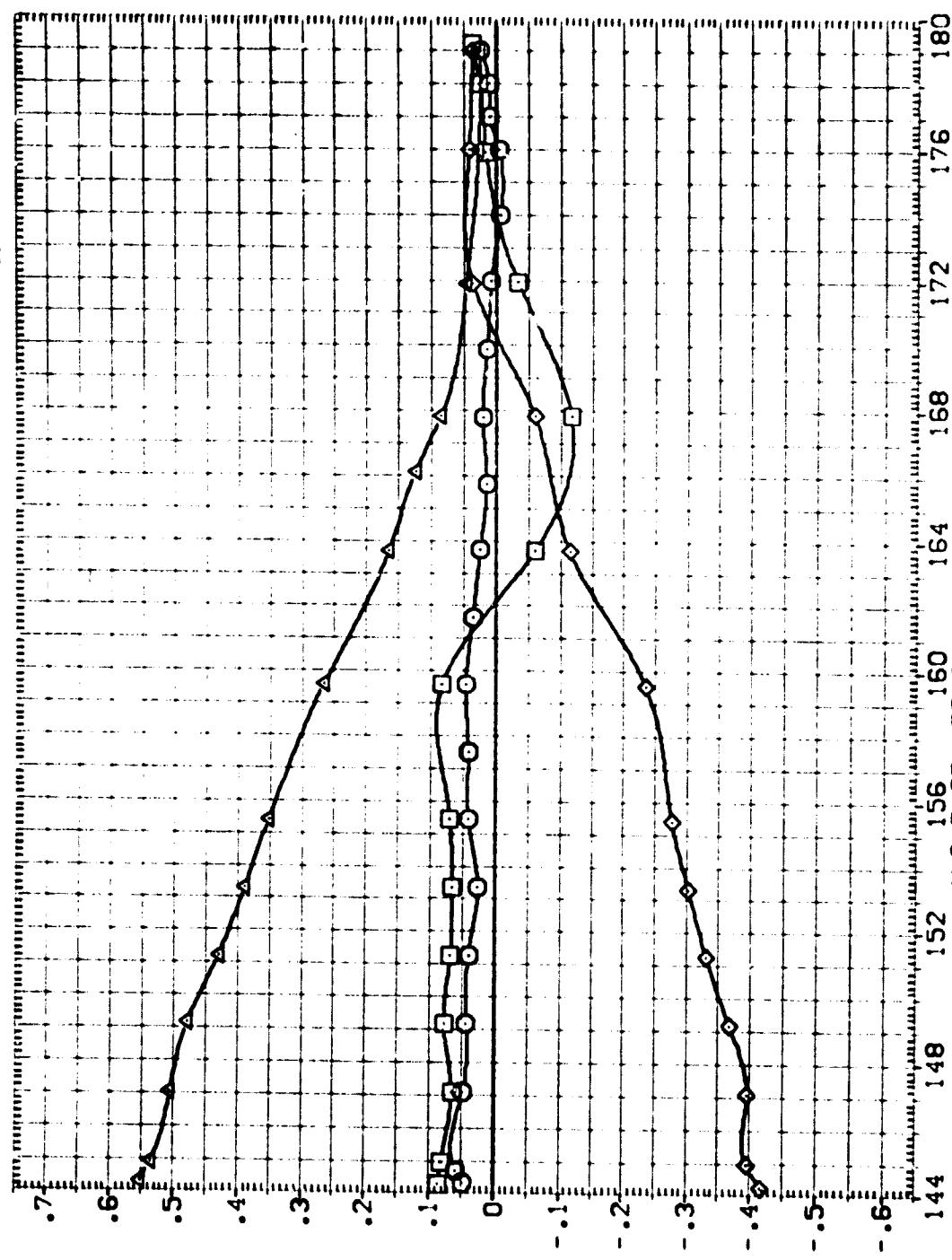
144 148 152 156 160 164 168 172 176 180
 EFFECT OF ROLL ANGLE ON STATIC STABILITY CHARACTERISTICS OF SRB

(C)MACH = 4.63

DATA SET NAME: CONFIGURATION DESCRIPTION
 [A-5001] SA-25F LARE UNIT 1087 MSFC 154
 [A-5002] SA-25F LARE UNIT 1087 MSFC 154
 [A-5003] SA-25F LARE UNIT 1087 MSFC 154
 [A-5004] SA-25F LARE UNIT 1087 MSFC 154

PHI : .000 .000 .000 .000
 TH ATTACH : 45.000 90.000 135.000
 ELT : TH ATTACH

REFERENCE INFORMATION
 SREF : .0690 SQ. IN.
 LREF : 3.0000 NOES
 BREF : 3.0000 NOES
 XMRP : 20.8340 NOES
 YMRP : .0000 NOES
 ZMRP : .0211 SCALE



MISSILE AXIS SIDE FORCE COEFFICIENT, CYM

EFFECT OF ROLL ANGLE ON STATIC STABILITY CHARACTERISTICS OF SRB
 (AJMACH = 2.70)

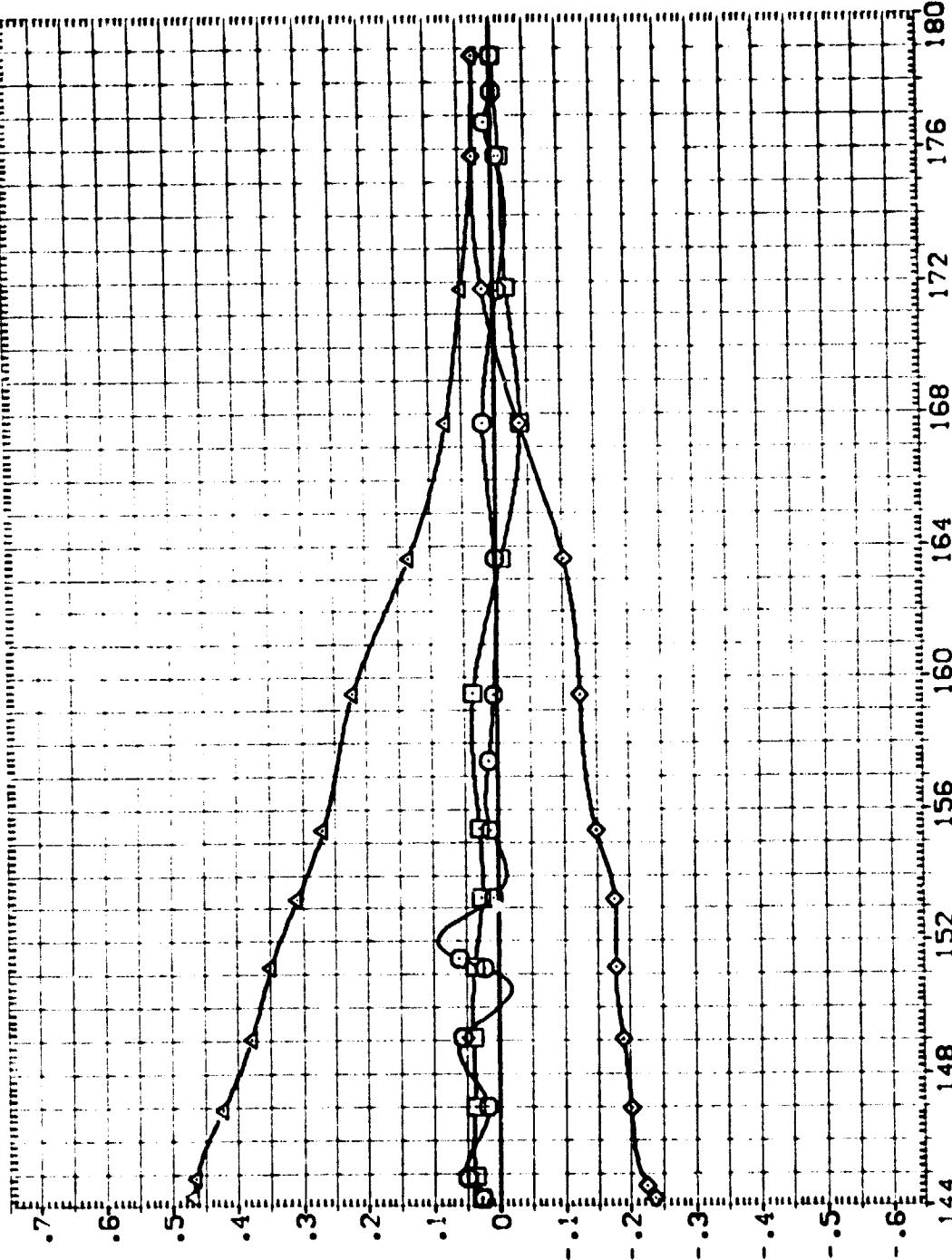
PAGE 16

DATA SET SYMBOL CONFIGURATION DESCRIPTION

| | | | | |
|----------|------------------|------|------|-----|
| [A-6001] | SA-25F LARC UNIT | 1087 | MSFC | 154 |
| [A-6002] | SA-25F LARC UNIT | 1087 | MSFC | 154 |
| [A-6003] | SA-25F LARC UNIT | 1087 | MSFC | 154 |
| [A-6004] | SA-25F LARC UNIT | 1087 | MSFC | 154 |

REFERENCE INFORMATION

| | | | |
|------|---------|--------|-------|
| SREF | .0680 | SD:IN: | .0000 |
| LREF | 3.0000 | INDESS | |
| BREF | 3.0000 | INDESS | |
| XMRP | 20.8040 | INDESS | |
| YMRP | .0000 | INDESS | |
| ZMRP | .0211 | SCALE | |



MISSILE AXIS SIDE FORCE COEFFICIENT, CYM

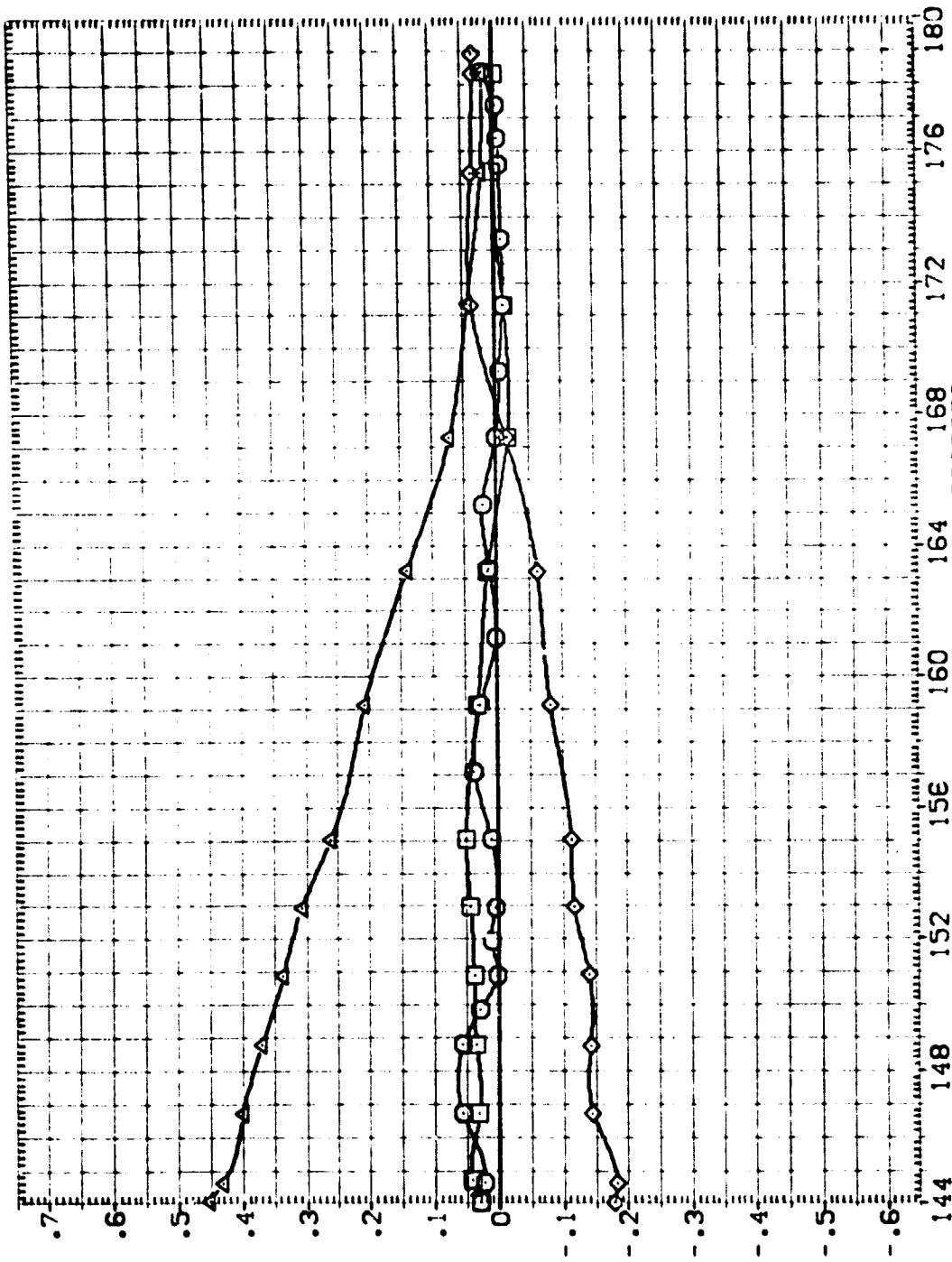
EFFECT OF ROLL ANGLE ON STATIC STABILITY CHARACTERISTICS OF SRB

(B)MACH = 4.00

DATA SET SPEED CONFIGURATION DESCRIPTION

| | | | | | | |
|-----------|--------|------|------|------|------|-----|
| [AH-5001] | SA-75F | LARC | UPNT | 1087 | MSFC | 454 |
| [RH-5002] | SA-75F | LARC | UPNT | 1087 | MSFC | 454 |
| [RH-5003] | SA-75F | LARC | UPNT | 1087 | MSFC | 454 |
| [RH-5004] | SA-75F | LARC | UPNT | 1087 | MSFC | 454 |

REFERENCE INFORMATION
 SREF 7.0680 SQ IN.
 LREF 3.0000 INCHES
 BREF 3.0000 INCHES
 XMAP 20.8340 INCHES
 YMAP .0000 INCHES
 ZMAP .0211 INCHES
 SCALE



MISSILE AXIS SIDE FORCE COEFFICIENT. CYM

EFFECT OF ROLL ANGLE ON STATIC STABILITY CHARACTERISTICS OF SRB

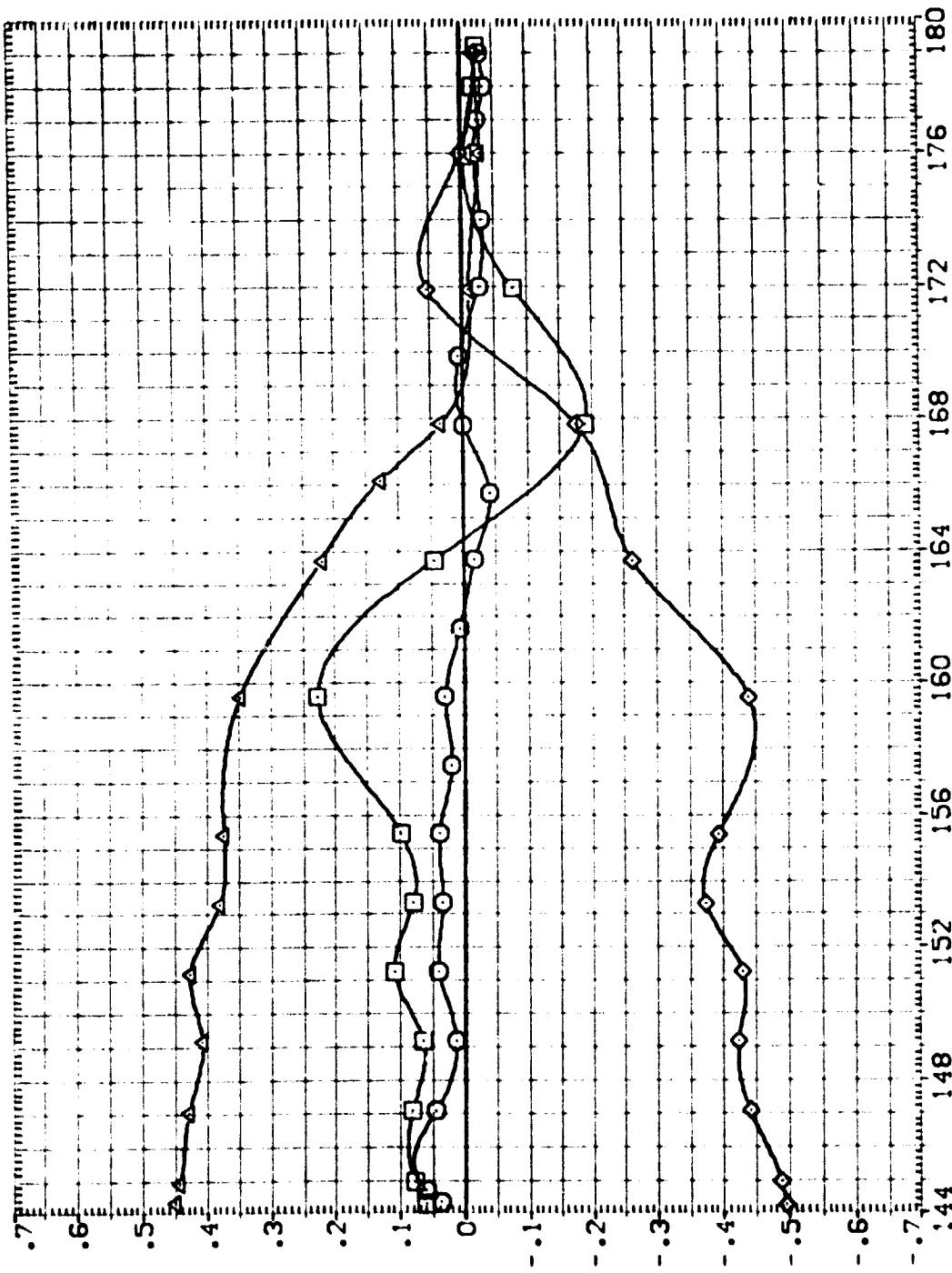
(C_{MACH} = 4.63)

PAGE 18

DATA SET NUMBER CONFIGURATION DESCRIPTION

| | | |
|----------|---|--------------------------------|
| [A-8001] | Q | SA-75F LARC UPNT 1087 MSEC 454 |
| [R-8002] | △ | SA-75F LARC UPNT 1087 MSEC 454 |
| [R-8003] | ○ | SA-75F LARC UPNT 1087 MSEC 454 |
| [R-8004] | × | SA-75F LARC UPNT 1087 MSEC 454 |

REFERENCE INFORMATION
 SREF 7.0690 SQ. IN.
 LREF 3.0000 INCHES
 BREF 3.0000 INCHES
 XMP 20.6930 INCHES
 YMP .0000 INCHES
 ZMP .0211 SCALE



MISSILE AXIS YAWING MOMENT COEFFICIENT, CYNM

EFFECT OF ROLL ANGLE ON STATIC STABILITY CHARACTERISTICS OF SRB

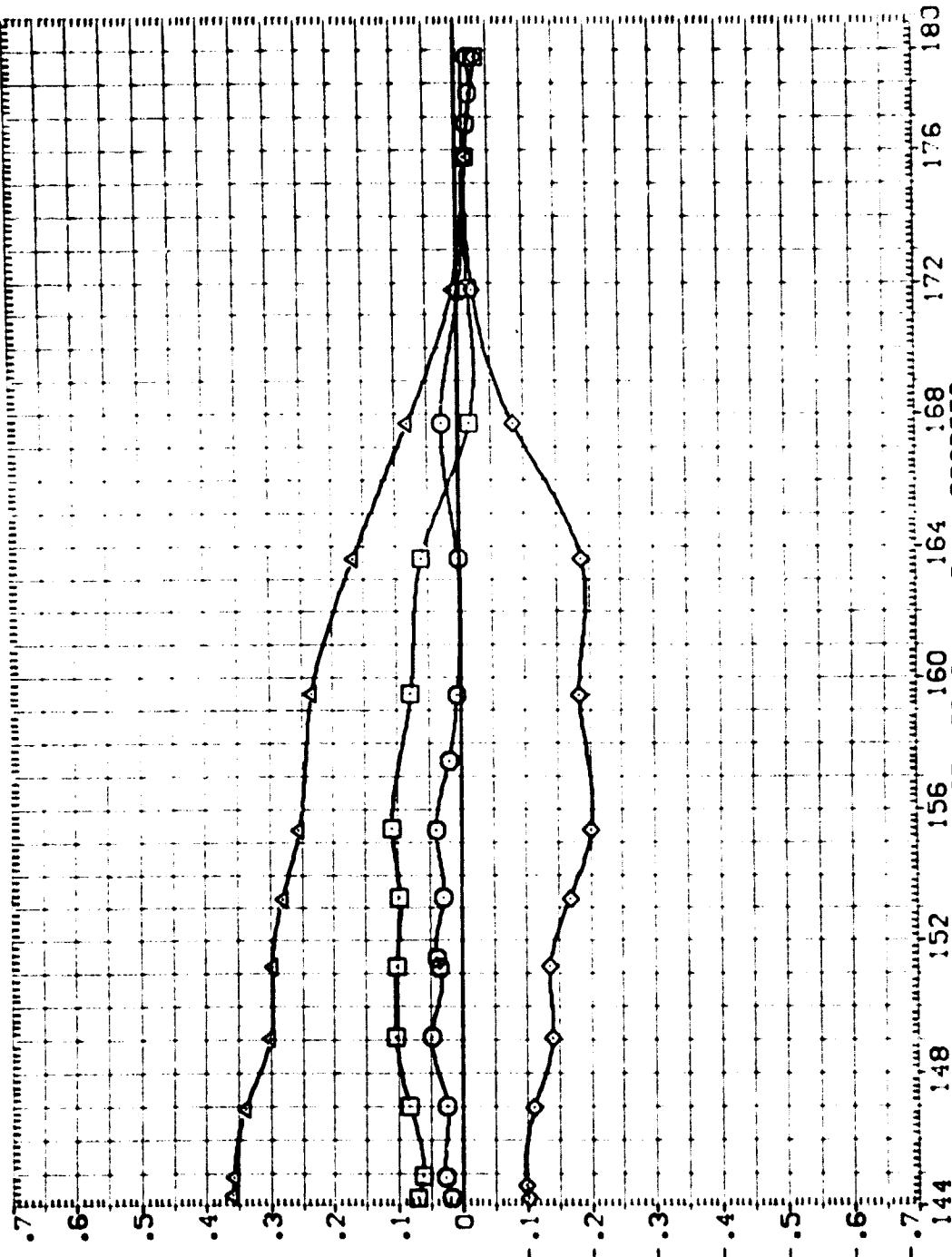
CAMMACH = 2.70

PAGE 19

DATA SET SYMBOL CONFIGURATION DESCRIPTION

| | | |
|----------|---|------------------|
| (A-8001) | □ | SA-75F LARC UPNT |
| (A-8002) | ○ | SA-75F LARC UPNT |
| (A-8003) | △ | SA-75F LARC UPNT |
| (A-8004) | △ | SA-75F LARC UPNT |

REFERENCE INFORMATION
 SREF 7.0690 SQ. IN.
 LREF 3.0000 INCHES
 BREF 3.0000 INCHES
 XREF 20.8340 INCHES
 YREF .0000 INCHES
 ZREF .0211 SCALE



MISSILE AXIS YAWING MOMENT COEFFICIENT. CYMN

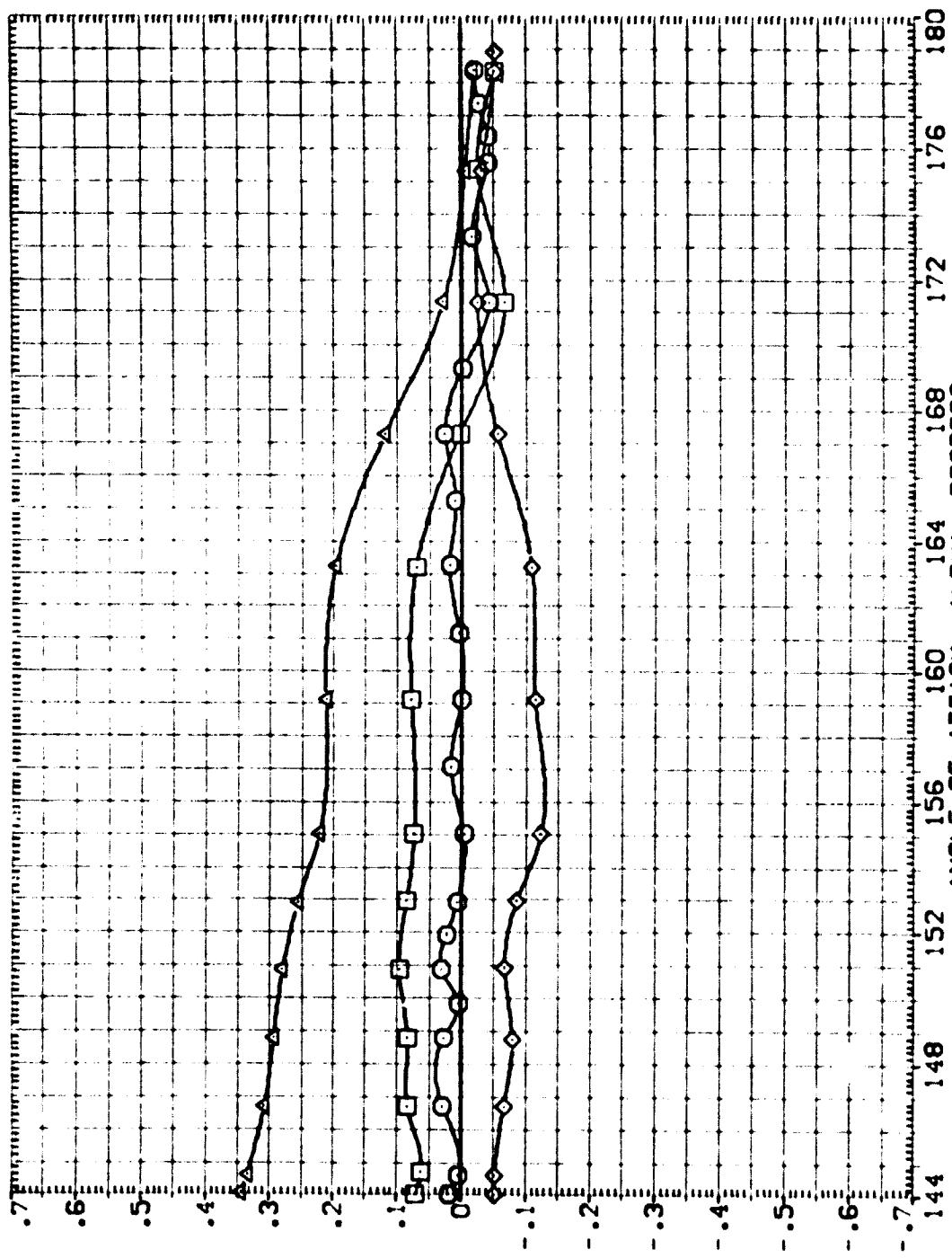
EFFECT OF ROLL ANGLE ON STATIC STABILITY CHARACTERISTICS OF SRB
 (B)MACH = 4.00

PAGE 20

DATA SET SYMBOL CONFIGURATION DESCRIPTION

| | | |
|----------|---|--------------------------------|
| [A-8001] | □ | SA-25F LARC UPNT 1087 NSFC 45A |
| [A-8002] | ○ | SA-25F LARC UPNT 1087 NSFC 45A |
| [A-8003] | × | SA-25F LARC UPNT 1087 NSFC 45A |
| [A-8004] | × | SA-25F LARC UPNT 1087 NSFC 45A |

REFERENCE INFORMATION
 SREF 7.0890 SD. IN.
 LREF 3.0000 NOES
 BREF 3.0000 NOES
 XMRP 20.8840 NOES
 YMRP .0000 NOES
 ZMRP .0000 NOES
 SCALE .0211



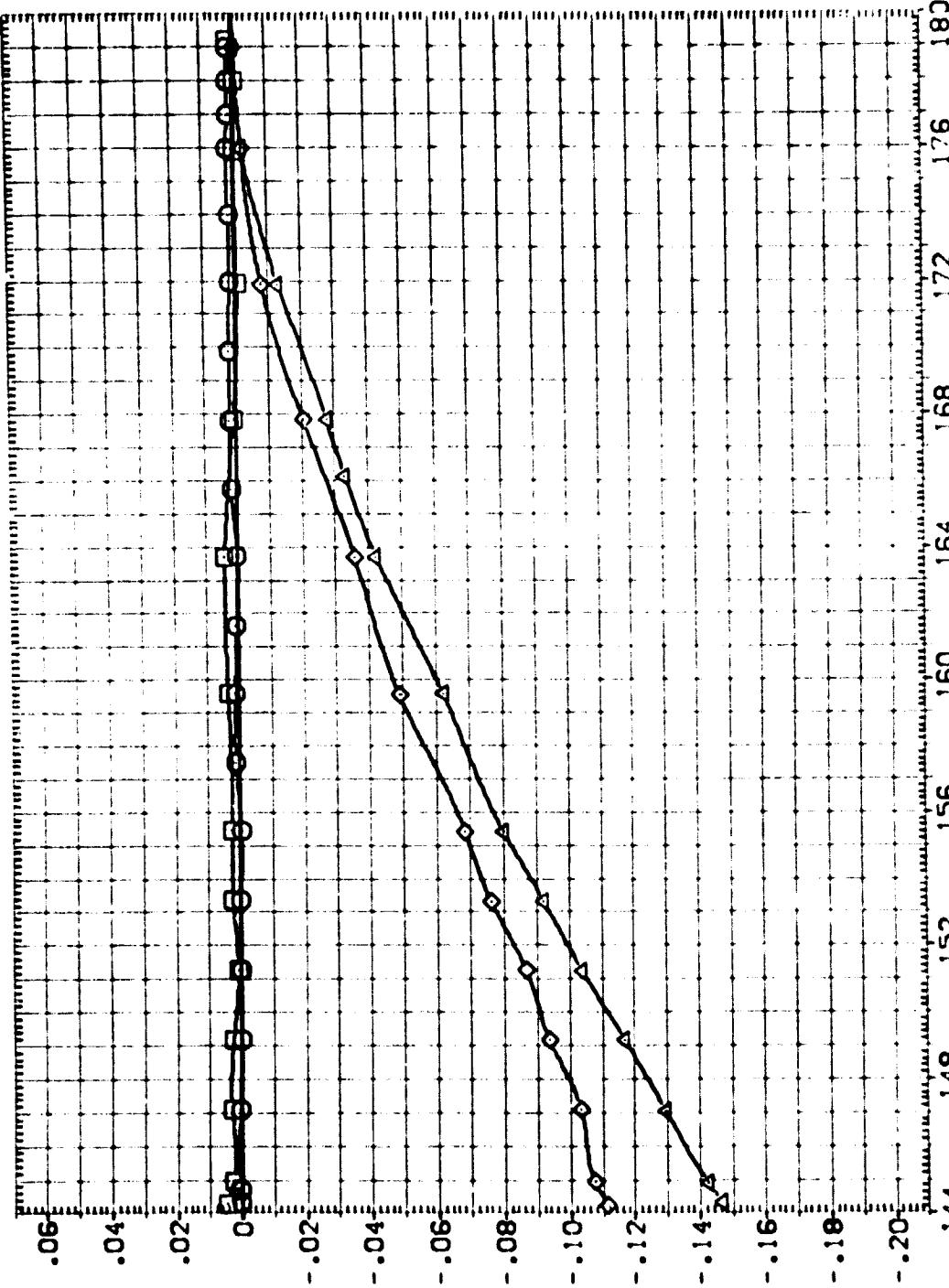
MISSILE AXIS YAWING MOMENT COEFFICIENT, CYNM

EFFECT OF ROLL ANGLE ON STATIC STABILITY CHARACTERISTICS OF SRB
 (C_{DA}) = 4.663

DATA SET SYMBOL CONFIGURATION DESCRIPTION

| | | | |
|--------|------------------|------|----------|
| AN0001 | SA-25F LANC UNIT | 1007 | MSFC 45° |
| AN0002 | SA-25F LANC UNIT | 1007 | MSFC 45° |
| AN0003 | SA-25F LANC UNIT | 1007 | MSFC 45° |
| AN0004 | SA-25F LANC UNIT | 1007 | MSFC 45° |

REFERENCE INFORMATION
 SREF 7.0650 SCALE .0211
 LREF 3.0000 INCL 13°
 BREF 3.0000 INCL 13°
 XHPP 20.8940 INCL 13°
 YHPP .0000 INCL 13°
 ZHPP .0000 INCL 13°
 SCALE .0211



MISSILE AXIS ROLLING MOMENT COEFFICIENT. CRL

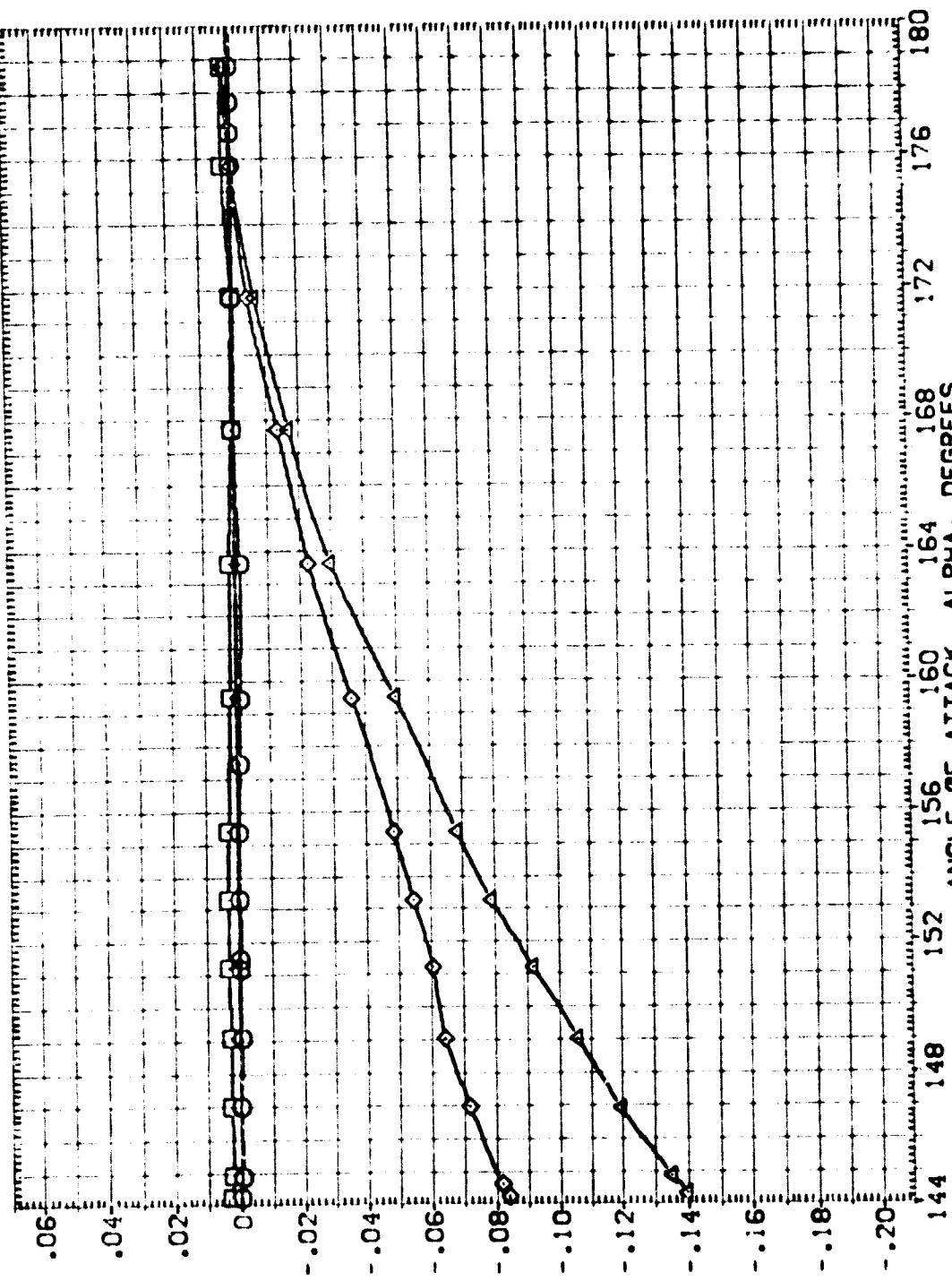
EFFECT OF ROLL ANGLE ON STATIC STABILITY CHARACTERISTICS OF SRB
 (C_AMACH = 2.70)

PAGE 22

DATA SET NUMBER CONFIGURATION DESCRIPTION

| | | | | | | |
|--------|--------|------|-----|------|------|-----|
| 148001 | SA-75F | LARE | UPN | 1087 | HSEC | 154 |
| 148002 | SA-75F | LARE | UPN | 1087 | HSEC | 454 |
| 148003 | SA-75F | LARE | UPN | 1087 | HSEC | 454 |
| 148004 | SA-75F | LARE | UPN | 1087 | HSEC | 454 |

REFERENCE INFORMATION
 SPREF 7.0630 60. IN.
 LREF 3.0000 INCHES
 BREF 3.0000 INCHES
 XMRP 20.6340 INCHES
 YMRP .0000 INCHES
 ZMRP .0000 INCHES
 SCALE .2211



MISSILE AXIS ROLLING MOMENT COEFFICIENT. CBL

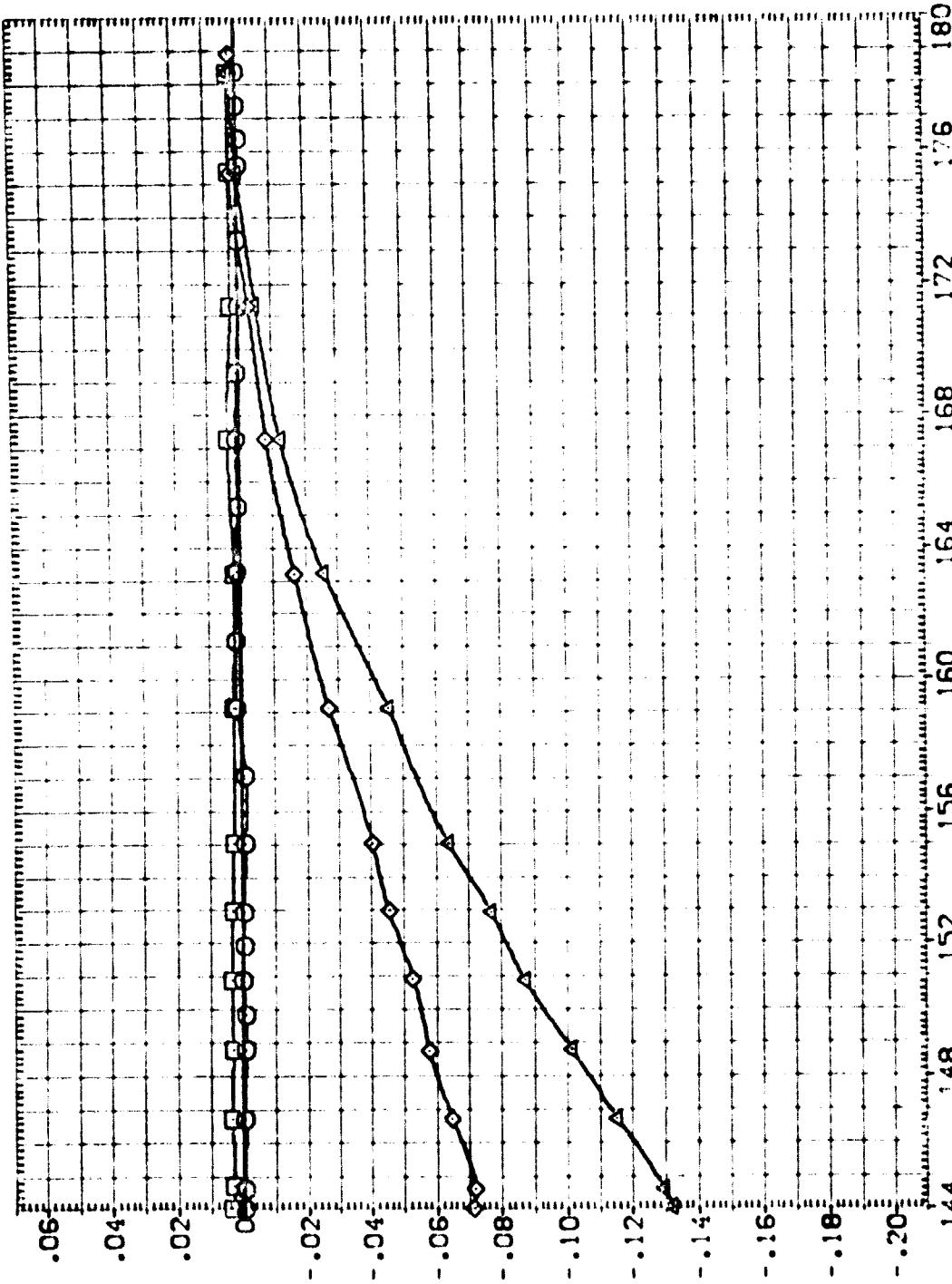
EFFECT OF ROLL ANGLE ON STATIC STABILITY CHARACTERISTICS OF SRB

(B)MACH = 4.00

DATA SET SYMBOL CONFIGURATION DESCRIPTION

| | |
|--------|-----------------|
| A-6001 | SA-255 LAR UNIT |
| B-6002 | 1087 HSC 154 |
| C-6003 | 1087 HSC 154 |
| D-6004 | 1087 HSC 154 |
| E-6005 | 1087 HSC 154 |

REFERENCE INFORMATION
 SREF 7.0690 SC:IN
 XREF 3.0000 INCHES
 YREF 3.0000 INCHES
 XHDP 2.8310 INCHES
 YHDP .0000 INCHES
 SCALE .0211



MISSILE AXIS ROLLING MOMENT COEFFICIENT, C_R

EFFECT OF ROLL ANGLE ON STATIC STABILITY CHARACTERISTICS OF SRB
 (C)_MACH = 4.63

APPENDIX

TABULATED SOURCE DATA

**Plotted data listings are available on request from
Data Management Services.**

011218 FEB 75

11801170 SOURCE DATA. LARC UPN 1087 MSC 494

3-22F LARC UPN 1087 MSC 494

PAGE 1

RHM9001) 14 FEB 75

REFERENCE DATA

BATT = 7.6000 SEC. IN.
LAMP = 20.0100 INCHES
LAMP = .00000 INCHES
BATT = 3.0000 INCHES
LAMP = .00000 INCHES
SCAL = .2211 SCALE

NUM NO. 1 / 0 MM/L = 1.00

GRADIENT INTERVAL = -5.00 / 5.00

| WAVE | WAVE | CMM | CA | CMM | CMM | CMM | CBL | CL | CD | CP(PS) |
|-------|---------|----------|----------|---------|----------|--------|--------|----------|-----------|-----------|
| 2.300 | 144.111 | 7.93871 | -2.92203 | .43946 | .07713 | .04770 | .00511 | -4.77565 | 6.93891 | 341.47734 |
| 2.300 | 144.465 | 7.93420 | -2.91932 | .73644 | .05211 | .03621 | .00503 | -6.64003 | 6.93232 | 341.71425 |
| 2.300 | 144.717 | 7.70140 | -2.91333 | .70236 | .05014 | .03570 | .00395 | -4.58457 | 6.83700 | 340.50008 |
| 2.300 | 145.055 | 6.95035 | -2.92084 | .48932 | .06110 | .02371 | .00360 | -6.25639 | 6.21705 | 340.50008 |
| 2.300 | 145.393 | 6.91674 | -2.92244 | .52259 | .06197 | .02529 | .00619 | -5.90619 | 5.76004 | 340.55201 |
| 2.300 | 151.029 | 5.69310 | -2.95054 | .12722 | .06339 | .02266 | .00359 | -3.54562 | 5.27623 | 340.50008 |
| 2.300 | 153.155 | 4.98735 | -2.86913 | -.01477 | .07698 | .01586 | .00461 | -3.13636 | 4.03118 | 340.61054 |
| 2.300 | 153.257 | 4.32633 | -2.85326 | -.01663 | .05965 | .01495 | .00626 | -2.13714 | 4.07232 | 340.54815 |
| 2.300 | 157.337 | 3.76150 | -2.81175 | -.36407 | .03234 | .02331 | .00448 | -2.31556 | 4.47215 | 340.52617 |
| 2.300 | 159.314 | 3.15864 | -2.77582 | -.15106 | .06310 | .02331 | .00344 | -1.98097 | 3.00263 | 340.26317 |
| 2.300 | 161.310 | 2.95223 | -2.72426 | -.69992 | .07470 | .02279 | .00507 | -1.69106 | 3.19811 | 340.38163 |
| 2.300 | 163.365 | 2.95503 | -2.65456 | -.82251 | .05813 | .01483 | .00486 | -1.69283 | 3.12552 | 340.3521 |
| 2.300 | 165.355 | 1.59757 | -2.62989 | -.91805 | .03981 | .01493 | .00433 | -1.69192 | 2.98152 | 340.32250 |
| 2.300 | 167.312 | 1.52014 | -2.56172 | -.90655 | .03513 | .01269 | .00454 | -1.5914 | 2.71124 | 340.29274 |
| 2.300 | 169.761 | 1.51423 | -2.51146 | -.83146 | .02276 | .00795 | .00339 | -1.39386 | 2.60568 | 340.44085 |
| 2.300 | 171.149 | 1.49751 | -2.47737 | -.74948 | .02101 | .00456 | .00695 | -1.11669 | 2.22187 | 340.36163 |
| 2.300 | 171.875 | 4.1046 | -2.46502 | -.67733 | .01978 | .00517 | .00470 | -0.66265 | 2.69463 | 340.26935 |
| 2.300 | 173.659 | 2.39162 | -2.39162 | -.50253 | .01906 | .00697 | .00271 | -0.1717 | 2.40404 | 340.04156 |
| 2.300 | 175.892 | 1.01720 | -2.31429 | -.31026 | .01706 | .00749 | .00311 | -0.06044 | 2.02332 | 340.73599 |
| 2.300 | 176.927 | .30701 | -2.27557 | -.26920 | .01746 | .00813 | .00350 | -0.03550 | 2.21697 | 339.99665 |
| 2.300 | 177.951 | .05073 | -2.23977 | -.18839 | .01909 | .01313 | .00632 | .01510 | 2.24065 | 340.26459 |
| 2.300 | 178.991 | .02048 | -2.17177 | -.07866 | .01313 | .00445 | .00607 | .01676 | 3.4100352 | |
| 2.300 | 179.443 | -.00238 | -2.16553 | -.03449 | .00988 | .00068 | .00000 | .00000 | | |
| | | GRADIENT | | | GRADIENT | | | | | |

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SA-2SF LARC UPUT 1007 MSFC 434

REFERENCE DATA

PARAMETRIC DATA

(RMS01) (14 FEB 73)

PAGE 2

REF. = 7.0000 30.1W.
 REF. = 3.0000 1W.
 REF. = 1.0000 1W.
 REF. = .33333 1W.
 REF. = .11111 1W.

| | REF. = 7.0000 30.1W. REF. = 3.0000 1W. REF. = 1.0000 1W. REF. = .33333 1W. REF. = .11111 1W. | REF. = 2.0000 1W. REF. = 1.0000 1W. REF. = .33333 1W. REF. = .11111 1W. | REF. = 1.0000 1W. REF. = .33333 1W. REF. = .11111 1W. | REF. = .33333 1W. REF. = .11111 1W. | REF. = .11111 1W. |
|---------------|--|--|---|--|-------------------|
| REF. = 7.000 | 1.0000 | 1.0000 | 1.0000 | 1.0000 | 1.0000 |
| REF. = 3.000 | .33333 | .33333 | .33333 | .33333 | .33333 |
| REF. = 1.000 | .11111 | .11111 | .11111 | .11111 | .11111 |
| REF. = .33333 | .03333 | .03333 | .03333 | .03333 | .03333 |
| REF. = .11111 | .01111 | .01111 | .01111 | .01111 | .01111 |
| REF. = .03333 | .00333 | .00333 | .00333 | .00333 | .00333 |
| REF. = .01111 | .00111 | .00111 | .00111 | .00111 | .00111 |
| REF. = .00333 | .00033 | .00033 | .00033 | .00033 | .00033 |
| REF. = .00111 | .00011 | .00011 | .00011 | .00011 | .00011 |
| REF. = .00033 | .00003 | .00003 | .00003 | .00003 | .00003 |
| REF. = .00011 | .00001 | .00001 | .00001 | .00001 | .00001 |
| REF. = .00003 | .00000 | .00000 | .00000 | .00000 | .00000 |
| REF. = .00001 | .00000 | .00000 | .00000 | .00000 | .00000 |
| REF. = .00000 | .00000 | .00000 | .00000 | .00000 | .00000 |



DATE 10 FEB 73

TABULATED SOURCE DATA - LARC UFWT 1087 (SA25F)

SA-25F LARC UFWT 1087 MSFC 454

PAGE 5

(RH9001) (14 FEB 73)

REFERENCE DATA

SATF = 7.0000 50.1IN. XWPF = 20.9340 INCHES
 LAFR = 3.0000 :INCHES VWF = .0000 INCHES
 BWF = 3.0000 INCHES ZWF = .0000 INCHES
 SCALE = .2211 SCALE

RUN NO. 1/0 RWF/L = 2.00 GRADIENT INTERVAL = -5.00/ 5.00

| MACH | ALPHA | CINN | CA | CLWN | CYWN | CBL | CL | CD | G (PSF) |
|-------|---------|----------|----------|---------|---------|---------|---------|----------|-------------------|
| 4.000 | 144.245 | 6.95902 | -2.68366 | .21336 | .02757 | .01887 | -.00025 | -4.00983 | 6.16849 290.54234 |
| 4.000 | 144.079 | 6.65399 | -2.65361 | .21160 | .02699 | .02699 | -.00041 | -3.91853 | 6.00351 290.58660 |
| 4.000 | 146.977 | 5.96409 | -2.61382 | .29585 | .01813 | .02458 | -.00018 | -3.57612 | 5.44186 290.59398 |
| 4.000 | 149.077 | 5.32017 | -2.59763 | .33559 | .03593 | .04652 | -.00035 | -3.24964 | 4.92053 290.54234 |
| 4.000 | 151.195 | 4.59972 | -3.00694 | 1.01713 | .02219 | .03503 | .00007 | -2.58121 | 4.85312 290.54234 |
| 4.000 | 151.455 | 4.52385 | -3.01332 | 1.02611 | .06129 | .03917 | -.00028 | -2.53304 | 4.80351 290.57923 |
| 4.000 | 153.397 | 4.02763 | -3.01778 | 1.04500 | .00996 | .02939 | .00002 | -2.22489 | 4.49644 290.19071 |
| 4.000 | 155.384 | 3.41784 | -2.93943 | 1.02134 | .01218 | .04015 | .00010 | -1.90105 | 4.10327 290.59398 |
| 4.000 | 157.455 | 2.92115 | -2.81474 | .99715 | .01467 | .01982 | -.00043 | -1.61038 | 3.73977 290.61611 |
| 4.000 | 159.481 | 2.47792 | -2.58123 | .72492 | .00396 | .00051 | -.00082 | -1.41597 | 3.28601 290.61611 |
| 4.000 | 153.612 | 1.37115 | -2.44140 | .47067 | .00245 | .00006 | -.00024 | -2.78550 | 290.61611 |
| 4.000 | 167.698 | .84573 | -2.32012 | .21203 | .01777 | .02816 | .00081 | -.33196 | 2.44724 290.63046 |
| 4.000 | 171.751 | .32955 | -2.24437 | .13840 | -.00587 | -.00239 | -.00228 | -.00412 | 2.25844 290.63024 |
| 4.000 | 175.790 | .10314 | -2.13555 | -.12637 | -.00547 | -.01247 | .00012 | .04357 | 2.12248 290.55299 |
| 4.000 | 176.605 | .07653 | -2.10869 | -.11685 | .00959 | -.01721 | -.00021 | .64110 | 2.11968 290.56691 |
| 4.000 | 177.739 | .05096 | -2.07073 | -.06754 | -.00395 | -.02188 | -.00013 | .93756 | 2.07119 290.56691 |
| 4.000 | 178.615 | .02491 | -2.05762 | -.03741 | -.00132 | -.02125 | -.00013 | .01766 | 2.05789 290.56691 |
| | | GRADIENT | .00000 | .00000 | .00000 | .00000 | .00000 | .00000 | .00000 |

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OF POOR QUALITY

SA-23F LARC UFLT 1007 MSFC 454

(RM9001) (114 FEB 73)

REFERENCE DATA

SREF = 7.0690 58.1IN. XMP = 20.0319 INCHES
 LEFF = 3.0000 INCHES YMP = .0000 INCHES
 BREP = 3.0000 INCHES ZMP = .0000 INCHES
 SCALE = .3211 SCALE

| RUN NO. | S/D | RN/L = | 2.00 | GRADIENT INTERVAL = -5.00/ 5.00 | | | |
|---------|----------|---------|-----------|---------------------------------|---------|---------|---------|
| | | | | C4M | C4M | C4M | CBL |
| 4.630 | ALPHA | 6.50320 | -2.59209 | .37361 | .02046 | .01896 | -.00040 |
| 4.630 | 144.924 | 6.40223 | -2.59232 | .45324 | .02224 | .00320 | -.00061 |
| 4.630 | 144.927 | 6.40223 | -2.59232 | .45324 | .02015 | .02015 | -.71967 |
| 4.630 | 146.719 | 5.75986 | -2.58112 | .51268 | .05536 | .01173 | -.00073 |
| 4.630 | 148.796 | 5.11830 | -2.48550 | .51941 | .05534 | .02728 | -.01988 |
| 4.630 | 149.930 | 4.84704 | -2.44987 | .52961 | .02919 | .00128 | -.00099 |
| 4.630 | 150.993 | 4.44639 | -2.87041 | 1.11099 | .00156 | .03240 | -.00002 |
| 4.630 | 151.951 | 4.14117 | -2.89075 | 1.08362 | .01094 | .02263 | -.00040 |
| 4.630 | 152.954 | 3.86713 | -2.88256 | 1.07495 | .00474 | .00711 | -.00062 |
| 4.630 | 155.926 | 3.35227 | -2.87740 | 1.06988 | .01112 | -.00153 | -.00098 |
| 4.630 | 157.059 | 2.03633 | -2.882436 | 1.03650 | .03643 | .01769 | -.00110 |
| 4.630 | 159.146 | 2.34756 | -2.71220 | .956677 | .02619 | .00096 | -.00138 |
| 4.630 | 161.176 | 1.94449 | -2.46710 | .72466 | .00214 | .00339 | -.00139 |
| 4.630 | 163.278 | 1.50824 | -2.49450 | .63114 | .01018 | -.01627 | .01311 |
| 4.630 | 165.246 | 1.13527 | -2.358081 | .45487 | .02150 | .00916 | -.0084 |
| 4.630 | 167.312 | .79785 | -2.31103 | .27769 | -.00067 | .02565 | -.01057 |
| 4.630 | 169.312 | .55620 | -2.28245 | .11174 | -.00656 | -.00155 | -.12270 |
| 4.630 | 171.329 | .34722 | -2.27083 | .01434 | -.01265 | -.04305 | -.00011 |
| 4.630 | 173.341 | .19427 | -2.21317 | -.19616 | -.01111 | -.01742 | .0006 |
| 4.630 | 175.370 | .09637 | -2.14472 | -.16346 | -.00861 | -.04492 | -.00033 |
| 4.630 | 176.395 | .06402 | -2.11293 | -.12588 | -.00715 | -.04140 | -.00033 |
| 4.630 | 177.396 | .06398 | -2.09200 | -.10307 | -.00375 | -.02725 | -.00115 |
| 4.630 | 178.389 | .03086 | -2.09037 | -.06669 | -.01556 | -.02069 | -.0024 |
| | GRADIENT | .00000 | .00000 | .00000 | .00000 | .00000 | .00000 |

PARAMETRIC DATA

(RMFSF)

5.96674 232.66627

5.62003 232.66627

5.29093 232.66627

4.77765 232.66627

4.53398 232.66627

4.67083 232.66627

4.48894 232.66627

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4.32574 232.66627

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1.32724 232.66627

1.12274 232.66627

.00000 232.66627



DATE 10 FEB 75

TABULATED SOURCE DATA - LARC UPNT 1007 (SA22F)

PAGE 7

SA-22F LARC UPNT 1007 MSFC 454 ELT TN ATTACH

(RH9002) (14 FEB 75)

REFERENCE DATA

SUPER = 7.0000 30-IN. XMP = 20.0340 INCHES
 LARP = 3.0000 INCHES YMP = .0000 INCHES
 BERP = 3.0000 INCHES ZMP = .0000 INCHES
 SCALE = .0211 SCALE

RUN NO. 7/0 RN/L = 1.50 GRADIENT INTERVAL = -5.00 / 5.00

| MACH | ALPHA | CMM | CA | CLMM | CYMM | CBL | CL | CD | (FSF) |
|----------|---------|---------|----------|---------|---------|---------|---------|-------------------|-------------------|
| 2.700 | 144.254 | 7.52603 | -2.61067 | .64704 | .08465 | .06009 | .00145 | -.4-.88307 | 6.51554 310.99829 |
| 2.700 | 144.935 | 7.29289 | -2.61368 | .62921 | .06134 | .07793 | .00257 | -.4-.16765 | 6.32910 310.89829 |
| 2.700 | 147.079 | 6.54775 | -2.96581 | 1.12153 | .06565 | .08106 | .00229 | -3.88392 | 6.04686 310.81254 |
| 2.700 | 149.177 | 5.49921 | -2.95553 | 1.00431 | .07795 | .05552 | .00236 | 5.56900 310.70103 | |
| 2.700 | 151.271 | 5.25545 | -2.94454 | .90825 | .06024 | .10543 | .00277 | -3.19341 | 5.10776 310.30652 |
| 2.700 | 153.359 | 4.63326 | -2.90653 | .78931 | .06161 | .07925 | .00228 | -2.84603 | 4.68191 310.81052 |
| 2.700 | 155.446 | 4.05581 | -2.85745 | .65219 | .07016 | .09876 | .00231 | -2.49272 | 4.28018 310.23460 |
| 2.700 | 159.590 | 2.91115 | -2.66479 | .22090 | .08276 | .22719 | .00347 | -1.88558 | 3.52666 310.14392 |
| 2.700 | 163.714 | 1.93977 | -2.53976 | -.11531 | -.05052 | .04700 | .00431 | 1.16890 | 2.98744 310.43994 |
| 2.700 | 167.853 | 1.01936 | -2.39570 | -.41427 | -.11672 | -.18993 | .00382 | -.48886 | 2.56830 310.28460 |
| 2.700 | 171.993 | .38235 | -2.41740 | -.49634 | -.03371 | -.07855 | -.00175 | -.04018 | 2.44772 310.35652 |
| 2.700 | 175.938 | .11985 | -2.25963 | -.32520 | -.11599 | -.00389 | -.00200 | -.00553 | 2.26244 311.16130 |
| 2.700 | 178.014 | .04656 | -2.15779 | -.21983 | -.02612 | -.01940 | -.00248 | -.00225 | 2.15811 311.16130 |
| 2.700 | 179.214 | -.00239 | -2.14823 | -.06077 | -.03222 | -.01515 | -.00187 | 2.14796 | 311.24967 |
| GRADIENT | | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 |

RUN NO. 9/0 RN/L = 2.00 GRADIENT INTERVAL = -5.00 / 5.00

| MACH | ALPHA | CMM | CA | CLMM | CYMM | CBL | CL | CD | (FSF) |
|----------|---------|---------|----------|---------|---------|---------|---------|-------------------|-------------------|
| 4.000 | 144.237 | 6.98219 | -2.67104 | .28116 | .08264 | .05600 | .00272 | -.4-.02355 | 8.11897 290.68687 |
| 4.000 | 144.973 | 6.55649 | -2.66473 | .31886 | .03725 | .06359 | .00252 | -3.91559 | 6.05082 290.68250 |
| 4.000 | 146.676 | 5.96223 | -2.65277 | .41097 | .03827 | .08143 | .00272 | -3.54569 | 5.44777 290.67512 |
| 4.000 | 151.407 | 5.31952 | -2.52651 | .42696 | .03943 | .10202 | .00290 | -3.23957 | 4.19436 290.66774 |
| 4.000 | 155.310 | 4.03085 | -2.52229 | 1.14476 | .02622 | .0947 | .00316 | 2.55557 290.60136 | |
| 4.000 | 159.210 | 1.582 | -2.56529 | 1.13930 | .02433 | .10266 | .00216 | 1.44215 | 4.22215 290.59223 |
| 4.000 | 163.155 | 1.13655 | -2.51317 | 1.13655 | .02613 | .09316 | .00308 | 1.99221 | 4.19464 290.59557 |
| 4.000 | 167.010 | .151 | -2.47882 | 2.47882 | .02161 | .07886 | .00216 | 1.44174 | 3.22877 290.58223 |
| 4.000 | 171.000 | .000 | -2.41694 | 2.41694 | .01717 | .05193 | .00193 | 1.44174 | 2.19031 290.58665 |
| 4.000 | 175.186 | -.07747 | -2.36195 | 2.36195 | .01148 | .03768 | .00193 | 1.44174 | 1.44174 290.58777 |
| 4.000 | 179.925 | -.00148 | -2.34919 | 2.34919 | .00639 | .02460 | .00193 | 1.44174 | 1.44174 290.58900 |
| GRADIENT | | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 |

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DATE 10 FEB 75

TABULATED SOURCE DATA - LARC UPWT 1987 (SA23F)

PAGE 8

SA-23F LARC UPWT 1987 MSFC 45° ELT TM ATTACH

(RM9002) (14 FEB 75)

REFERENCE DATA

SREF = 7.0690 30.1IN.
 LREF = 3.0000 INCHES THRP = 29.8340 INCHES
 GREF = 3.0000 INCHES THRP = .0000 INCHES
 SCALER = .0211 SCALE

PARAMETRIC DATA

| RUN NO. | 0 / 0 | RNL = 2.01 | GRADIENT INTERVAL = -.500 / .500 | | | | | | | |
|---------|----------|------------|----------------------------------|---------|---------|---------|--------|----------|---------|-----------|
| MACH | A - PMA | CMM | CA | CLMM | CTM | CYMM | CBL | CL | CD | O (PSF) |
| 4.630 | 144.030 | 6.59330 | -2.61974 | .51660 | .02930 | .07044 | .00310 | -3.79900 | 5.99495 | 233.50038 |
| 4.630 | 144.693 | 6.41336 | -2.61402 | .59739 | .04214 | .06218 | .00287 | -3.72110 | 5.83998 | 233.50038 |
| 4.630 | 146.757 | 5.77335 | -2.51962 | .69430 | .03056 | .06217 | .00319 | -3.42774 | 5.30855 | 233.50038 |
| 4.630 | 148.800 | 5.12413 | -2.49773 | .69213 | .03517 | .08169 | .00300 | -3.09758 | 4.79607 | 233.50038 |
| 4.630 | 150.866 | 4.62531 | -2.08056 | 1.3050 | .03131 | .06160 | .00317 | -2.49719 | 4.68787 | 233.50038 |
| 4.630 | 152.966 | 3.66341 | -2.88763 | 1.18472 | .03453 | .08551 | .00293 | -2.14383 | 4.35812 | 233.50038 |
| 4.630 | 155.040 | 3.37236 | -2.87616 | 1.19819 | .04984 | .07419 | .00259 | -1.84370 | 4.03064 | 233.50038 |
| 4.630 | 159.149 | 2.33951 | -2.73153 | 1.06592 | .03032 | .07832 | .00248 | -1.21015 | 3.38536 | 233.50038 |
| 4.630 | 163.199 | 1.50225 | -2.41407 | .59120 | .01488 | .07153 | .00197 | -7.74356 | 2.45524 | 233.50038 |
| 4.630 | 167.313 | .79562 | -2.32979 | .29273 | .01887 | .00302 | .00373 | -2.26520 | 2.13885 | 233.50038 |
| 4.630 | 171.321 | .34593 | -2.27696 | -.02206 | -.01365 | -.05727 | .00235 | .00162 | 2.30309 | 233.50038 |
| 4.630 | 175.375 | .09599 | -2.15951 | -.18848 | -.00109 | -.02558 | .00257 | .07446 | 2.16032 | 233.50038 |
| 4.630 | 179.352 | -.00110 | -2.09905 | -.07730 | -.00267 | -.05267 | .00226 | .06415 | 2.09015 | 233.50038 |
| | GRADIENT | .00009 | .00000 | .00000 | .00000 | .00000 | .00000 | .00000 | .00000 | .00000 |



DATE 10 MAY 78

TABULATED SOURCE DATA, LANC UPW 1987 (SAP95)

5
page

REFERENCE DATA

| SCALE | .0211 SCALE | 1.0000 36 IN. | 1MM | 2 | 20.0140 INCHES |
|--------|-------------|---------------|-----|---|----------------|
| LARGE | 2 | 3.0000 INCHES | 1MM | 2 | .0555 INCHES |
| MEDIUM | 3 | 2.0000 INCHES | 2MM | 2 | .0500 INCHES |
| SMALL | 4 | 1.0000 INCHES | 2MM | 2 | .0500 INCHES |

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GRADIENT INTERVAL = -3.02 / 3.00

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EAGANETICS DATA

SI-29E LANE LIGHT 1002 MILE 111 EAT ON ATTACH

MEMO 111 EER 74

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EAGANETICS DATA

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MIN NO. 10/ 0 RNU = 2.00 GRADIENT INTERVAL = -5.00 / 5.00

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SA-25F LARC UPWT 1007 MSFC 454 ELT TA11CH

(14 FEB 75)

REFERENCE DATA

| | |
|-------|----------------|
| SREF | 7.0000 30.1IN. |
| LREF | 3.0000 INCHES |
| BREF | 3.0000 INCHES |
| SCALF | .20211 SCALE |

PARAMETRIC DATA

FH1 = 90.000

FH2 = 90.000

RUN NO. 11/0 RNL = 2.01 GRADIENT INTERVAL = -5.00/ 5.00

| INCN | ALPHA | CIN | CA | CLMN | CYN | CYNH | CBL | CL | CD | CFSP |
|-------|---------|----------|----------|---------|----------|----------|----------|----------|---------|-----------|
| 4.630 | 144.005 | 6.79122 | -2.61263 | .57256 | -1.67085 | -0.05162 | -0.0125 | -3.95012 | 6.10509 | 233.54011 |
| 4.630 | 144.613 | 5.61177 | -2.60201 | .62269 | -1.6266 | -0.05323 | -0.0149 | -3.95018 | 5.95018 | 233.54011 |
| 4.630 | 145.752 | 5.94063 | -2.55230 | .63963 | -1.55555 | -0.06531 | -0.0171 | -3.5496 | 5.35002 | 233.54011 |
| 4.630 | 146.769 | 5.31059 | -2.49763 | .65016 | -1.47111 | -0.0749 | -0.0257 | -3.25229 | 4.89326 | 233.54011 |
| 4.630 | 146.769 | 4.59389 | -2.60152 | 1.25221 | -1.3666 | -0.06703 | -0.0252 | -2.61574 | 4.75025 | 233.54011 |
| 4.630 | 146.938 | 4.04313 | -2.69987 | 1.1766 | -1.1756 | -0.0652 | -0.0231 | -2.28316 | 4.42058 | 233.54011 |
| 4.630 | 155.011 | 3.48608 | -2.65979 | 1.12419 | -1.13114 | -0.0632 | -0.0219 | -1.94563 | 4.02376 | 233.54011 |
| 4.630 | 156.139 | 2.44685 | -2.72237 | 1.05699 | -0.8155 | -0.0615 | -0.0192 | -0.52693 | 3.21121 | 233.54011 |
| 4.630 | 163.210 | 1.59129 | -2.41666 | .61016 | -0.6237 | -0.05163 | -0.0156 | -0.82356 | 2.71331 | 233.54011 |
| 4.630 | 167.280 | .90359 | -2.31979 | .29288 | -0.1630 | -0.05565 | -0.00823 | -3.55117 | 2.45729 | 233.54011 |
| 4.630 | 171.326 | .34919 | -2.27947 | .00882 | .01666 | -0.0213 | -0.00140 | -2.30606 | 2.30606 | 233.54011 |
| 4.630 | 175.335 | .10312 | -2.16402 | -1.1884 | .03485 | -0.0355 | .006125 | 2.16573 | 2.16573 | 233.54011 |
| 4.630 | 176.347 | .02677 | -2.06669 | .03183 | .03183 | -0.03187 | .00384 | 2.10559 | 2.10559 | 233.54011 |
| 4.630 | 176.926 | .02959 | -2.09087 | .05766 | .05766 | -0.05193 | .0067 | 2.06106 | 2.06106 | 233.54011 |
| | | .00000 | | .00000 | | | | | | |
| | | GRADIENT | | | | | | | | |



34-259 LABC INPUT 1087 MSFC 454 ELT IN ATTACH

(R40904) (10 FEB 75)

REFERENCE DATA

| REF | 7.0000 30.1IN. | 1INP = | 20.0340 INCHES | RH1 = | 135.000 |
|--------|----------------|--------|----------------|-------|---------|
| LREF | 3.0000 INCHES | VINP = | .00000 INCHES | | |
| BREF | 1.0000 INCHES | 2INP = | .00000 INCHES | | |
| SCALE: | .0211 SCALE | | | | |

PARAMETRIC DATA

| | RUN NO. | 145/0 RHWL = 1.50 | GRADIENT INTERVAL = -5.00/ 5.00 | | |
|-------|---------|-------------------|---------------------------------|---------|--------|
| MACH | ALPHA | CIN | CA | CIN | CIN |
| 2.700 | 146.327 | 7.45603 | -2.37700 | 0.1680 | .59294 |
| 2.700 | 146.314 | 7.27231 | -2.39347 | 0.1700 | .51607 |
| 2.700 | 147.041 | 6.53219 | -2.93997 | .58966 | .50800 |
| 2.700 | 149.100 | 5.60688 | -2.84556 | .48227 | .50962 |
| 2.700 | 151.213 | 5.23969 | -2.95502 | .36515 | .52396 |
| 2.700 | 153.326 | 4.61153 | -2.95205 | .20473 | .53262 |
| 2.700 | 155.412 | 4.00039 | -2.84950 | .21964 | .51252 |
| 2.700 | 167.816 | 1.01242 | -2.49071 | .58636 | .08522 |
| 2.700 | 171.320 | .38025 | -2.39775 | .51094 | .01261 |
| 2.700 | 173.365 | .13074 | -2.32775 | .32260 | .02181 |
| 2.700 | 179.021 | .04447 | -2.3193 | .02113 | .01926 |
| | | .00000 | | .00000 | .00000 |
| | | | | | |
| | RUN NO. | 155/0 RHWL = 2.00 | GRADIENT INTERVAL = -5.00/ 5.00 | | |
| MACH | ALPHA | CIN | CA | CIN | CIN |
| 4.000 | 144.310 | 6.99398 | -2.65384 | 0.20064 | .49562 |
| 4.000 | 144.937 | 6.72560 | -2.65317 | 0.20460 | .49562 |
| 4.000 | 146.927 | 6.02548 | -2.69601 | 0.22207 | .33696 |
| 4.000 | 149.045 | 5.33392 | -2.65376 | 0.30575 | .37982 |
| 4.000 | 151.206 | 4.61771 | -3.01315 | .66151 | .36699 |
| 4.000 | 153.292 | 1.93158 | -2.49195 | .59638 | .24149 |
| 4.000 | 155.412 | .38025 | -2.49071 | .58636 | .01261 |
| 4.000 | 167.816 | .13074 | -2.39775 | .51094 | .02181 |
| 4.000 | 171.320 | .04447 | -2.32775 | .32260 | .01926 |
| 4.000 | 173.365 | .00000 | | .00000 | .00000 |
| | | | | | |
| | RUN NO. | 155/0 RHWL = 2.00 | GRADIENT INTERVAL | | |
| MACH | ALPHA | CIN | CA | CIN | CIN |
| 4.000 | 144.310 | 6.99398 | -2.65384 | 0.20064 | .49562 |
| 4.000 | 144.937 | 6.72560 | -2.65317 | 0.20460 | .49562 |
| 4.000 | 146.927 | 6.02548 | -2.69601 | 0.22207 | .33696 |
| 4.000 | 149.045 | 5.33392 | -2.65376 | 0.30575 | .37982 |
| 4.000 | 151.206 | 4.61771 | -3.01315 | .66151 | .36699 |
| 4.000 | 153.292 | 1.93158 | -2.49195 | .59638 | .24149 |
| 4.000 | 155.412 | .38025 | -2.49071 | .58636 | .01261 |
| 4.000 | 167.816 | .13074 | -2.39775 | .51094 | .02181 |
| 4.000 | 171.320 | .04447 | -2.32775 | .32260 | .01926 |
| 4.000 | 173.365 | .00000 | | .00000 | .00000 |

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DATE 08 098 78

INSTRUMENT SOURCE DATA. LAMP UNIT 1007 MSCP 446 ELET TH ATTACH

(NH9504) 114 FEB 75 1

REFERENCE DATA

| | REF ID | REF ID | REF ID | REF ID | |
|--------|-------------------------------|--------|-------------------------------|--------|-------------------------------|
| REF ID | 3.99999 100000 0.00000 100000 | REF ID | 3.99999 100000 0.00000 100000 | REF ID | 3.99999 100000 0.00000 100000 |
| REF ID | 3.99999 100000 0.00000 100000 | REF ID | 3.99999 100000 0.00000 100000 | REF ID | 3.99999 100000 0.00000 100000 |
| REF ID | 3.99999 100000 0.00000 100000 | REF ID | 3.99999 100000 0.00000 100000 | REF ID | 3.99999 100000 0.00000 100000 |
| REF ID | 3.99999 100000 0.00000 100000 | REF ID | 3.99999 100000 0.00000 100000 | REF ID | 3.99999 100000 0.00000 100000 |

BIN NO. 1860 MUL = 2.00 COVENT INTERVAL = -3.00 / 5.00

RFL = 135.000

PARAMETRIC DATA

3.99999 100000 0.00000 100000 ELET TH ATTACH

REF ID

(NH9504) 114 FEB 75 1

REF ID

DATE 10 FEB 79

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TABLED SOURCE DATA. LARC UPRT 1007 (10257)

SA-25F LARC UPRT 1007 MSFC 454 ELT TH ATTACH

(FM9002) (14 FEB 79)

REFERENCE DATA

SURF = 7.5000 36.1IN.
LREV = 3.00000 INCHES
SERV = 3.00000 INCHES
SCALE = 0211 SCALE

RUN NO. 7/ D RNL = 1.30 GRADIENT INTERVAL = -5.00/ 5.00

| MACH | ALPHA | CPC1 | CPC2 | 0 (PSF) |
|-------|----------|--------|--------|-----------|
| 2.700 | 144.254 | -13213 | -13276 | 310.89129 |
| 2.700 | 144.935 | -13213 | -13584 | 310.89129 |
| 2.700 | 147.979 | -13519 | -13891 | 310.89124 |
| 2.700 | 149.177 | -13316 | -13580 | 310.70103 |
| 2.700 | 151.271 | -13291 | -13572 | 310.30152 |
| 2.700 | 153.369 | -13211 | -13274 | 310.81052 |
| 2.700 | 155.469 | -12853 | -13264 | 310.26167 |
| 2.700 | 159.390 | -12598 | -12659 | 310.43012 |
| 2.700 | 163.714 | -12231 | -12343 | 310.45954 |
| 2.700 | 167.653 | -11682 | -12030 | 310.26160 |
| 2.700 | 171.953 | -09617 | -10180 | 310.29052 |
| 2.700 | 175.938 | -07389 | -07746 | 311.16130 |
| 2.700 | 178.914 | -06775 | -07131 | 311.16130 |
| 2.700 | 179.214 | -06412 | -06827 | 311.24697 |
| | GRADIENT | .00000 | .00000 | |

RUN NO. 9/ D RNL = 2.00 GRADIENT INTERVAL = -5.00/ 5.00

| MACH | ALPHA | CPC1 | CPC2 | 0 (PSF) |
|-------|----------|--------|--------|-----------|
| 4.000 | 144.237 | -03713 | -03815 | 290.68987 |
| 4.000 | 144.903 | -03713 | -04144 | 290.68250 |
| 4.000 | 146.976 | -04022 | -04144 | 290.67512 |
| 4.000 | 149.062 | -04409 | -04473 | 290.66174 |
| 4.000 | 151.210 | -04727 | -04801 | 290.60136 |
| 4.000 | 153.310 | -05365 | -05789 | 290.62148 |
| 4.000 | 155.407 | -05364 | -05789 | 290.57923 |
| 4.000 | 159.494 | -05714 | -06119 | 290.60250 |
| 4.000 | 163.599 | -05713 | -05789 | 290.58680 |
| 4.000 | 167.722 | -05365 | -05460 | 290.61011 |
| 4.000 | 171.850 | -04723 | -04787 | 290.32195 |
| 4.000 | 175.766 | -06665 | -04667 | 290.27079 |
| 4.000 | 179.803 | -03738 | -04139 | 290.36631 |
| | GRADIENT | .00000 | .00000 | |

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SA-214 L488 C4987 WRE 1454 ELT TM ATTACH

LHM(24) (4457 74)

REFERENCE DATA

| | SLN NO. | 14/5 | SLN NO. | 14/5 | SLN NO. | 14/5 |
|-------|---------|---------------|---------|----------------|---------|------|
| SPEP | | 7-2000 INCHES | | 20-0346 INCHES | | |
| Lamp | | 3.9000 INCHES | | 3.9000 INCHES | | |
| BSP | | 3.9000 INCHES | | 3.9000 INCHES | | |
| SCALF | | 3.9111 SCALE | | 3.9111 SCALE | | |

SLN NO. 14/5 8MM = 2.500 GRADIENT INTERVAL 7 - 3.900 / 3.900

| W4CH | W-5HA | CPC1 | CPC2 | G(F3P) |
|---------|---------|---------|----------|--------|
| 144.579 | -0.2384 | -0.2376 | 232.9756 | |
| 144.524 | -0.2364 | -0.2376 | 232.9756 | |
| 144.524 | -0.2304 | -0.2376 | 232.9756 | |
| 146.599 | -0.2317 | -0.2396 | 232.9756 | |
| 146.599 | -0.2317 | -0.2396 | 232.9756 | |
| 148.535 | -0.2384 | -0.2396 | 232.9756 | |
| 148.535 | -0.2384 | -0.2396 | 232.9756 | |
| 150.509 | -0.2364 | -0.2396 | 232.9756 | |
| 152.555 | -0.2323 | -0.2396 | 232.9756 | |
| 152.555 | -0.2323 | -0.2396 | 232.9756 | |
| 155.516 | -0.2413 | -0.2419 | 232.9756 | |
| 155.516 | -0.2413 | -0.2419 | 232.9756 | |
| 158.540 | -0.2413 | -0.2429 | 232.9756 | |
| 159.505 | -0.2432 | -0.2429 | 232.9756 | |
| 163.521 | -0.2533 | -0.2429 | 232.9756 | |
| 165.524 | -0.2624 | -0.2429 | 232.9756 | |
| 167.569 | -0.2723 | -0.2429 | 232.9756 | |
| 171.543 | -0.3223 | -0.2429 | 232.9756 | |
| 175.552 | -0.3244 | -0.3337 | 232.9756 | |
| 178.565 | -0.3244 | -0.3337 | 232.9756 | |
| 180.550 | .00000 | .00000 | .00000 | |

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